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OGDEN AIR LOGISTICS CENTER HILL AFB UT PROPELLANT LAB--ETC F/G 21/9.2  
PROPELLANT SURVEILLANCE REPORT LGM-30 A, B, F & G STAGE I TP-HI--ETC(U)  
OCT 81 J A THOMPSON

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OGDEN AIR LOGISTICS CENTER  
UNITED STATES AIR FORCE

PROPELLANT  
SURVEILLANCE REPORT  
LGM-30A, B, F&G STAGE I  
TP-H 1043

PROPELLANT LAB SECTION

MANPA REPORT

NR 462(81)

OCTOBER 1981

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(6) PROPELLANT SURVEILLANCE REPORT  
LGM-30 A, B, F & G STAGE I  
TP-H1043 AFT CLOSURE PROPELLANT

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## ABSTRACT

This report contains propellant test results from cartons of TP-H1043 propellant representing selected batches used in the aft closure of First Stage Minuteman Motors. Data from TP-H1043 propellant obtained from the aft closures of the LCM-30 A, B, F, and G Motors are reported in regression analyses for the sixth time and the fifth time using the G085 Computer System. In previous reports, A, B, F, and G data were combined for statistical analysis. For this report, a separate statistical analysis was made for A and B (wing 2) and F and G (wing 6). The statistical analysis includes regressions and covariance analysis. Testing was accomplished in accordance with MMWRBA Project M8293C.

An analysis of all parameters indicate that no significant degradation is anticipated for at least two years past the oldest data point.

Each point on the regression plot represents all samples at that particular age. The number of samples at each point is indicated on the sample size summary sheet on the page accompanying each regression plot. The data range at any age can be found by suitable inquiry of the G085 System.

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# LIST OF REFERENCES

<u>Report Nr</u>	<u>Title</u>	<u>Report Date</u>
	LGM-30 First Stage, Wing I Test Reports	
29D	Zero Time Test Results (Aft Closure)	9 Jun 64
29E	Zero Time (Aft Closure Supplement 1)	24 Jun 64
32B	Zero Time, Wings II-V Test Results (Aft Closure)	18 Mar 65
185	ATP Phase I, Wing VI Series III	Jun 70
195	ATP Phase III, Wing II-V	Nov 70
239	Propellant Surveillance Report (TP-H1043)	Apr 72
288	Propellant Surveillance Report (TP-H1043)	Mar 74
337	Propellant Surveillance Report (TP-H1043)	Feb 76
385	Propellant Surveillance Report (TP-H1043)	Dec 77
424	Propellant Surveillance Report (TP-H1043)	Nov 79



# GLOSSARY OF TERMS AND ABBREVIATIONS

Aging Trend	A change in properties or performance resulting from aging of material or component
CSA	Cross Sectional Area
DB	Dogbone
Degradation	Gradual deterioration of properties or performance
E	Modulus (psi), defined as stress divided by strain along the initial linear portion of the curve
EB	End Bonded
EGL	Effective Gage Length
em	Strain at maximum stress
er	Strain at rupture
"F" ratio	The ratio of the variance accounted for by the regression function to the random unexplained variance. The regression function having the most significant "F" ratio is used for plotting data. The ratio is also used in detecting significant changes in random variation between succeeding time points.
JANNAF	Joint Army, Navy, NASA, Air Force Committee
MAGCP	Propellant Lab Section at OOAMA
OOAMA	Ogden Air Materiel Area, Air Force Logistics Command
Regression Equation	The general form of the regression equation is $Y = a + bx$
Regression Line	Line representing mean test values with respect to time
$S_b$	Standard error of estimate of the regression coefficient
$S_e$ or $S_{Y.X}$	Standard deviation of the data about the regression line

# GLOSSARY OF TERMS AND ABBREVIATIONS (cont)

SM	Maximum Stress
Sr	Stress at rupture
Standard Deviation(S )	Square root of variance
Strain Rate	Crosshead speed divided by the EGL
"t" test	A statistical test used to detect significant differences between a measured parameter and an expected value of the parameter (determines if regression slope differs from zero at the 95% confidence level)
Variance	The sum of squares of deviations of the test results from the mean of the series after division by one less than the total number of test results
3 Sigma Band	The area between the upper and lower 3 sigma limit. It can be expected that 99.73% of the inventory represented by the test samples would fall within this range assuming that the population is normally distributed.
90-90 Band	It can be stated with 90% confidence that 90% of the inventory represented by the test samples would fall within this range assuming that the population is normally distributed.

## INTRODUCTION

### A. PURPOSE:

Quality assurance tests have been conducted for 14 1/2 years on First Stage LGM-30 TP-H1043 aft closure propellant.

Statistical analysis of the tests performed, as directed by Engineering, should provide early warning if serious degradation trends occur. Evaluation of the propellant provides data that can be put directly into engineering reliability and service life predictions. Testing was performed in accordance with MMWRBA Directive GTD-1C, Amendments 1 and 2.

### B. BACKGROUND:

TP-H1043 propellant is used in the aft closure of the LGM-30 A, B, F, and G First Stage Motors.

This test period represents the sixth time that TP-H1043 propellant has been reported by regression analysis. This is also the fifth time that data has been processed utilizing the G085 System.

### C. SAMPLING PLAN:

As many as four aft closures are cast from the one TP-H1043 propellant mix. In order to reduce the number of tests, only one batch from each mix will be tested to obtain uniform test results. The selected batches are from the same batch as those previously tested and reported in MANCP Reports 185(70), 195(70), 239(72), 288(74), 385(77), and 424(79).

Low rate tensile, high rate tensile and hardness tests were performed on each propellant batch mix.

## STATISTICAL ANALYSIS

Analysis of covariance was selected as a method of statistically comparing regression trend lines from wing 2 and wing 6 data. Although both groups of data are from the same population, they have undergone unknown biasing and will be handled separately. This was discovered during the analysis of covariance test. Therefore, these two groups of data will no longer be statistically placed in the same regressions.

Linear regression analysis was selected as the method for data evaluation. Data from different time periods were used to establish a least squares trend line for the data. The variance about the regression line, obtained using individual values of the dependent variable, was used to compute a tolerance interval such that at the 90% confidence level, 90% of the sample distribution fall within this interval. This tolerance interval was extrapolated to a maximum of 24 months. The "t" values and the significance of this statistic, which are reported for each regression model, give an indication of the "statistical significance" of the slope of the trend line as compared to a line of zero slope.

Each point on the regression analysis is a calculation of all samples at that particular age. The number of samples at each point is indicated on the sample size summary sheet accompanying each regression plot. The data range at any age can be found by suitable inquiry of the G085 System.

The analysis of covariance uses lower experimental error and a more precise comparison among treatments (wings). The regressions may differ in slope direction, elevation, or in residual variances. The most convenient approach is to compare the residual variance first, then the

slopes, and lastly the elevations. The residual mean squares are compared by the two-tailed F-test at the 5% significant level. The null hypothesis that Wing 2 and Wing 6 population residual variances are statistically equal must be rejected in all comparisons. The scatter of the points about the individual regression lines suggests that the differences in slope and elevation may be attributed to sampling variations.

The slight difference in slope and trend directions found in Wing 2 and Wing 6 data are mostly due to differences in sample quantities and number of test periods represented. However, all regression slopes are statistically close to a linear line of zero slope regardless of the direction. Additional Wing 6 testing should provide the required number of samples which should provide compatibility with the Wing 2 test results.

## TEST RESULTS

### A. LOW RATE TENSILE:

The strains at maximum stress for wing two and wing six regressions (Fig 1 and 2), show a statistically significant decreasing slope. The analysis of covariance (Table 2), shows a statistically significant difference in variance and elevation with no significant difference in slope.

For maximum stress the wing two regression shows a statistically decreasing slope and the wing six regression shows a non-significant slope (Fig 3 and 4). The analysis of covariance shows a statistically significant difference for variance, slope, and elevation (Table 3).

The strain at rupture, for wing 2, shows a statistically significant increasing slope with a statistically significant decreasing slope for the wing six regression (Fig 5 and 6). The analysis of covariance shows a statistically significant difference for variance, slope, and elevation (Table 4).

The stress at rupture regression for wing two shows a statistically significant decreasing slope with a non-significant slope for wing six (Fig 7 and 8). The analysis of covariance shows a statistically significant difference for variance, slope, and elevation (Table 5).

Modulus regressions show a statistically significant decreasing slope for wing two and a non-significant slope for wing six (Fig 9 and 10). The analysis for covariance shows a statistically significant difference for variance, slope, and elevation (Table 6).

### B. HIGH RATE TENSILE:

The strain at maximum stress regression for wing two shows a statistically significant decreasing slope and the wing six regression shows a statistically significant increasing slope (Fig 11 and 12). The analysis of covariance shows a statistically significant difference for variance, slope, and elevation (Table 7).

The maximum stress regression shows a statistically significant decreasing slope, with wing six regression showing a statistically significant increasing slope (Fig 13 and 14). The analysis for covariance shows a statistically significant difference for variance and slope with no significant difference for elevation (Table 8).

Strain at rupture regressions show a statistically significant decreasing slope for wing two and a statistically significant increasing slope for wing six (Fig 15 and 16). The analysis for covariance shows a statistically significant difference for variance, slope, and elevation (Table 9).

Stress at rupture regressions shows a statistically non-significant slope for wing two and wing six (Fig 17 and 18). The analysis for covariance shows a statistically significant difference for variance, slope, and elevation (Table 10).

Modulus regressions show a statistically significant increasing slope for wing two and a statistically significant decreasing slope for wing six (Fig 19 and 20). The analysis for covariance shows a statistically significant difference for variance, slope, and elevation (Table 10).

#### C. HARDNESS:

Shore A initial hardness regressions for wing two and wing six show statistically significant increasing slopes (Fig 21 and 22). The analysis for covariance shows a statistically significant difference for variance, slope, and elevation (Table 11).

Shore A ten second hardness regressions show a statistically significant increasing slope for wing two and a statistically significant decreasing slope for wing six (Fig 23 and 24). The analysis for covariance

shows a statistically significant difference for variance and slope with no difference for elevation (Table 12).

Shore C initial hardness regression shows a statistically significant decreasing slope for wing two and wing six (Fig 25 and 26). Analysis for covariance shows a statistically significant difference for variance and elevation with no difference for slope (Table 13).

Shore C ten second hardness regressions show a statistically significant decreasing slope (Fig 27 and 28). Analysis for covariance shows a statistically significant difference for variance, slope, and elevation (Table 14).



## CONCLUSIONS AND RECOMMENDATIONS

Wing two and wing six data cannot be combined for statistical analysis.

For both wing two and wing six, the slopes of the regressions are gradual and close to a line of zero slope. From this analysis, the propellant degradation is gradual and the propellant service life may be extended for at least two years beyond the date of the last testing.

It is recommended that testing be continued to assure service life extension and confirm present trends.

TABLE 1  
ANALYSIS OF COVARIANCE  
COMPARISON OF REGRESSIONS AT THE 5 % SIGNIFICANT LEVEL

** TYPE **		** REGRESSIONS **		
TEST		VARIANCE	SLOPE	ELEVATION
TENSILE -- LOW RATE --	strain at max stress	S.	N.S.	S.
	max stress	S.	S.	S.
	strain at rupture	S.	S.	S.
	stress at rupture	S.	S.	S.
	modulus	S.	S.	S.
	-- HIGH RATE --			
	strain at max stress	S.	S.	S.
	max stress	S.	S.	N.S.
	strain at rupture	S.	S.	S.
HARDNESS -- SHORE A --	stress at rupture	S.	S.	S.
	modulus	S.	S.	S.
	initial	S.	S.	S.
	10-sec.	S.	S.	N.S.
-- SHORE C --				
	initial	S.	N.S.	S.
	10-sec.	S.	S.	S.

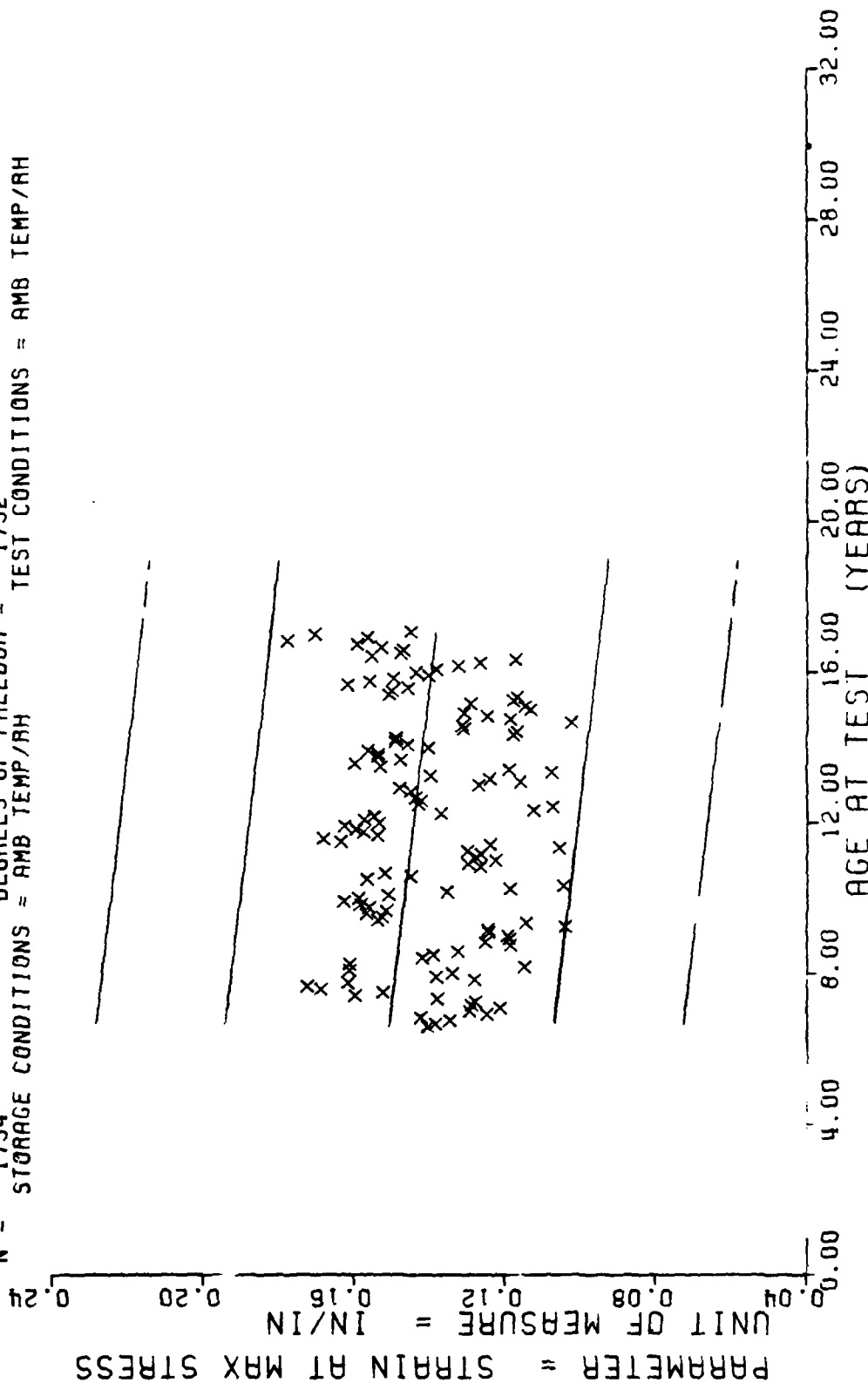
N.S.=Not significantly different  
S. =Significantly different

[illegible]

STAGE 1, 416.2, TD-H1043, STAIN AT MAX GTP45, LOW RATE, CHS=2.0

This sample size summary is applicable to figure 1, 3, 5, 7, and 9

$Y = ((+1.5842743E-01) + (-9.8519768E-05) * X)$   
 $F = +4.2054089E+01$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma_r = +2.6238808E-02$   
 $R = -1.5310393E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_e = +1.5192150E-05$   
 $L = +6.4849124E+00$  SIGNIFICANCE OF L = SIGNIFICANT  $S_t = +2.5936854E-02$   
 $N = 1754$  DEGREES OF FREEDOM = 1752  
 STORAGE CONDITIONS = AMB TEMP/ RH TEST CONDITIONS = AMB TEMP/ RH



STAGE 1. WING 2. IP-H1043. STRAIN AT MAX STRESS. LOW RATE. CHS=2.0

Figure 1

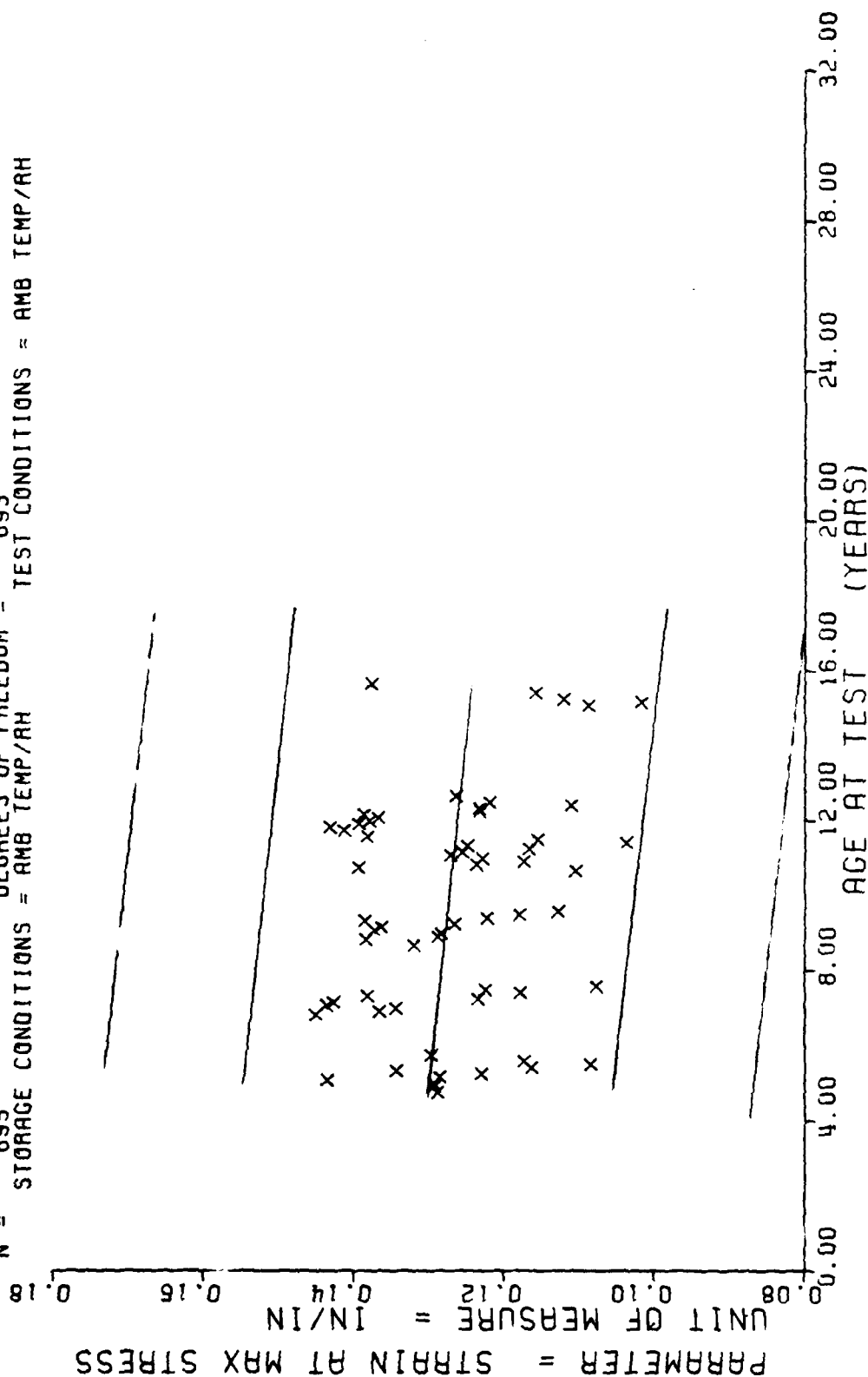
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
57	1	106	24	147	14		
58	0	110	18	148	15		
60	3	111	27	149	12		
61	6	112	16	150	5		
62	0	117	24	152	3		
63	6	116	14	161	5		
64	3	115	9	162	3		
65	0	120	3	163	1		
66	9	129	3	165	2		
67	0	130	6	168	3		
69	3	171	15				
72	6	132	26				
73	6	133	20				
74	30	134	30				
75	12	135	17				
76	0	136	18				
77	15	137	21				
78	3	138	18				
79	18	139	3				
80	3	141	3				
81	3	142	6				
104	2	143	30				
106	3	144	20				
107	12	145	18				
108	21	146	24				

STAGE 1, WING 1, 12-H1043, LIPAD AT TAX SHEDS, LOW PATH, CHS=2.0

This sample size summary is applicable to figure 2, 4, 6, 8, and 10

$F = +5.4767541E+00$  SIGNIFICANCE OF F = (-4.5849890E-05) \* X)  
 $R = -8.8549404E-02$  SIGNIFICANT  
 $t = +2.3402466E+00$  SIGNIFICANT  
 $N = 695$  DEGREES OF FREEDOM = 693  
 STORAGE CONDITIONS = AMB TEMP/AH TEST CONDITIONS = AMB TEMP/AH





$Y = ((+1.9440499E+02) + (-2.1261750E-01) * X)$   
 F = +1.4901312E+02      SIGNIFICANCE OF F = SIGNIFICANT       $\sigma^2 = +3.0974085E+01$   
 R = -2.7990167E-01      SIGNIFICANCE OF R = SIGNIFICANT       $S_b = +1.7417537E-02$   
 t = +1.2207093E+01      SIGNIFICANCE OF t = SIGNIFICANT       $S_e = +2.9744490E+01$   
 N = 1755      DEGREES OF FREEDOM = 1753  
 STORAGE CONDITIONS = AMB TEMP/AH      TEST CONDITIONS = AMB TEMP/AH

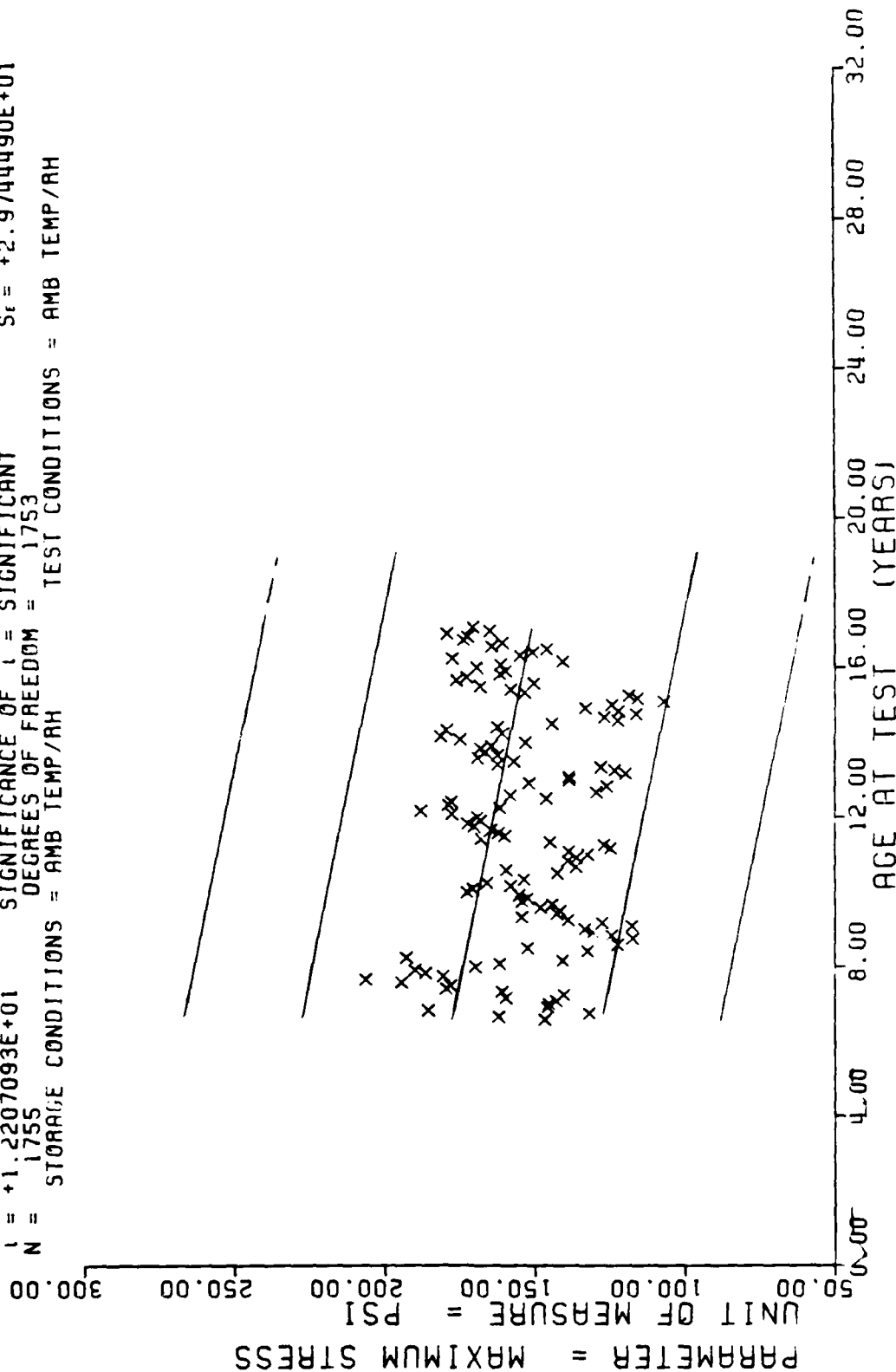






TABLE 3

ANALYSIS OF COVARIANCE  
COMPARISON OF REGRESSION LINES

ANALYSIS OF COVARIANCE COMPARES REGRESSIONS, TP-HIGH, CFE=2.0 MAX STRESS, KING-2/B

SOURCE	D.F.	SUM XX	CORRELATED SUM SQUARES	SUM XY	SUM YX	COEFFICIENTS	D.F.	DEVIATIONS FROM REGRESSION	SUM SQ.	MEAN SQ.
ADJUSTED										
AXS2(1)	1754	0.29163E+07	-0.62007E+06	0.14033E+07	-0.21262		1753	0.15509E+07	864.74	
AXS6(1)	694	0.59233E+05	-1.2497	0.59233E+05	-0.41474E-01		693	0.32130E+06	460.71	
TOTAL	2448	0.30587E+07	-0.64256E+06	0.20051E+07	-0.18570		2447	0.16657E+07	770.61	
DIFFERENCE BETWEEN SLOPES	1						1	13592.	13592.	
REJECTED	1	77056.	66656.	37406.						
TOTAL	2449	1.35357E+07	-0.57591E+06	0.16027E+07			2448	0.15985E+07		
DELETED ADJUSTED SQUARES	1						1	63.52.	63.52.	

STEP 1:

COMPARISON BETWEEN RESIDUAL VARIABLES : F = 1.91 (DF=1753, 692)

IF COEFFICIENCY OF RESIDUAL VARIABLES IS IDENTICAL WITH STEP 2.

STEP 2:

COMPARISON OF REGRESSION TREND LINE SLOPES : F = 17.90 (DF = 1, 2446)

COMPARISON OF REGRESSION TREND LINE DEVIATIONS : F = 180.05 (DF = 1, 2447)

STATISTICALLY DIFF.

STATISTICALLY DIFF.  
STATISTICALLY DIFF.

$Y = ((+2.7902915E-01) + (+5.6128604E-05) * X)$   
 $F = +4.3818054E+00$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma_1 = +4.5822020E-02$   
 $R = +4.9947882E-02$  SIGNIFICANCE OF R = SIGNIFICANT  $S_0 = +2.6813758E-05$   
 $t = +2.0932762E+00$  SIGNIFICANCE OF t = SIGNIFICANT  $S_t = +4.5777885E-02$   
 $N = 1754$  DEGREES OF FREEDOM = 1752  
 STORAGE CONDITIONS = AMB TEMP/AH TEST CONDITIONS = AMB TEMP/RH

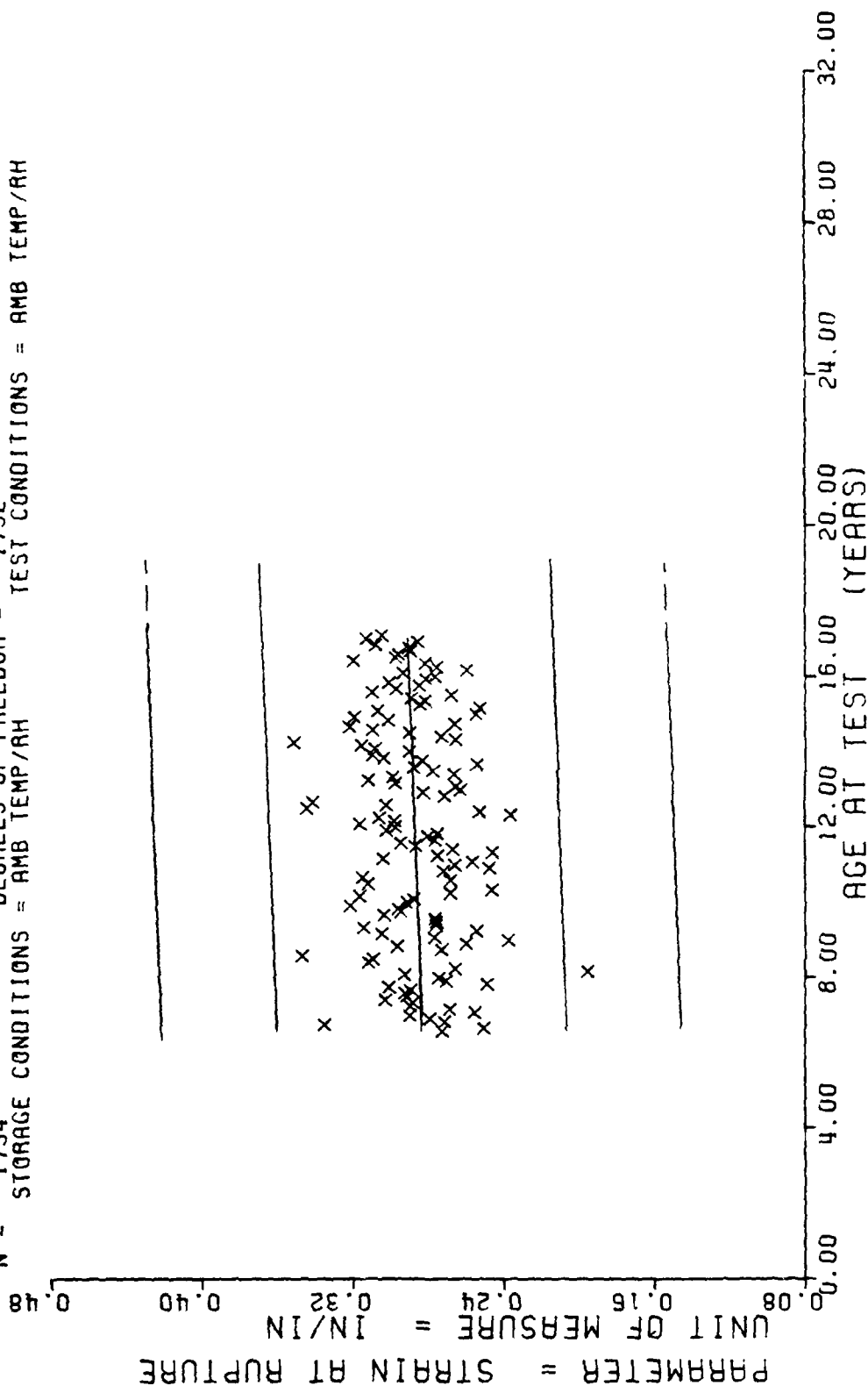
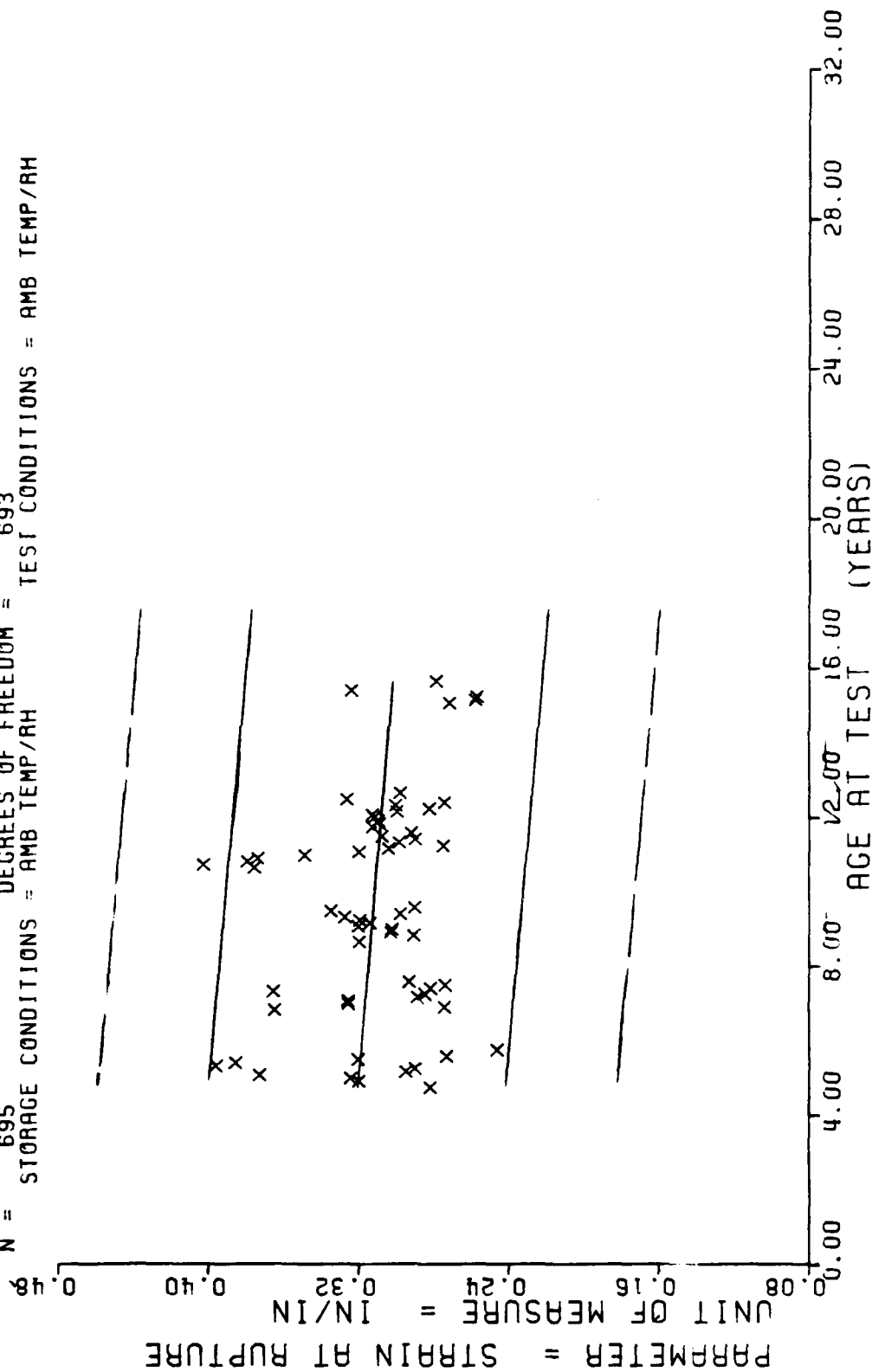


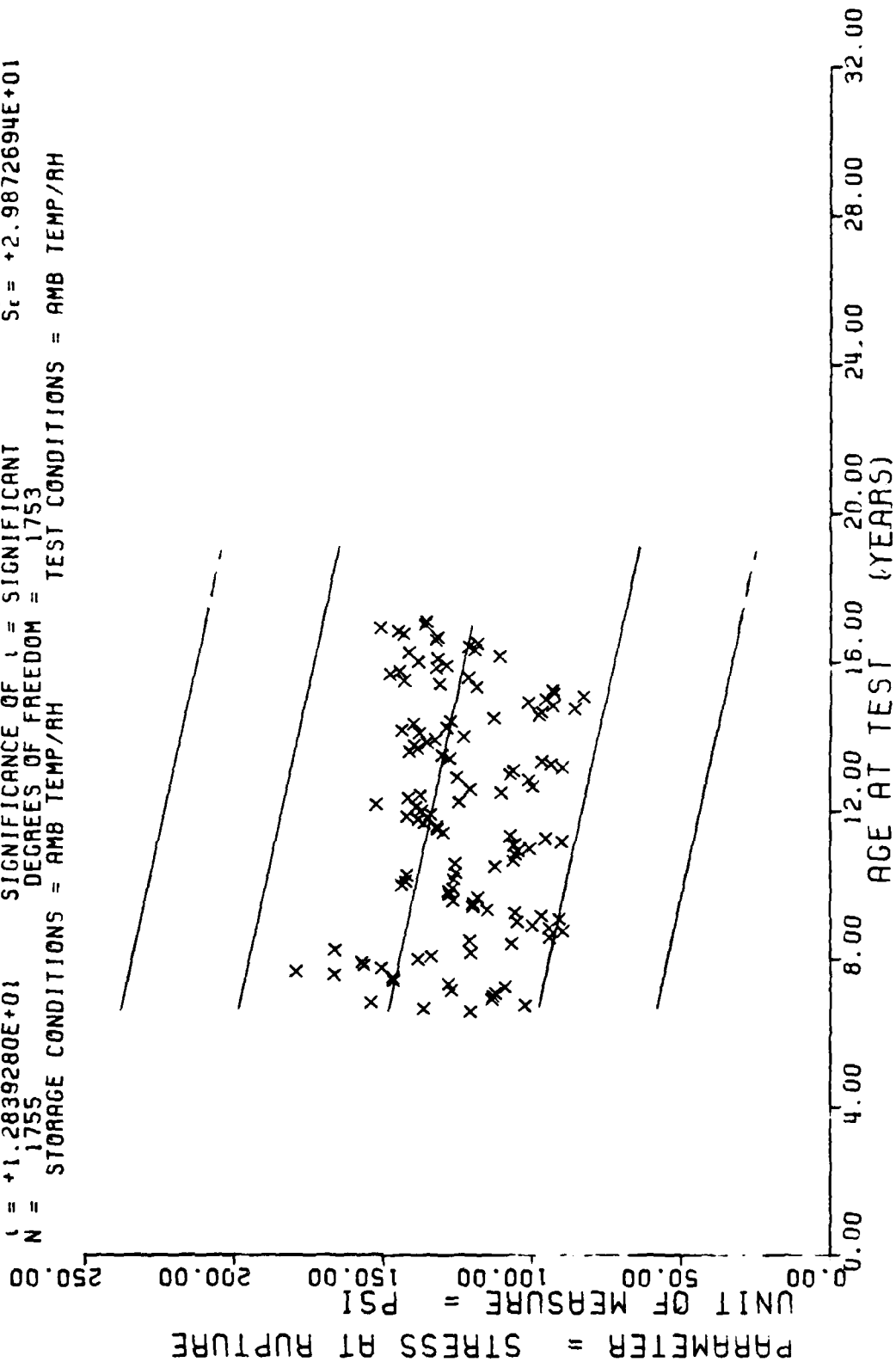
Figure 5

$Y = ((+3.2906807E-01) + (-1.4962635E-04) \times X)$   
 $F = +5.7256720E+00$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma_r = +4.6205637E-02$   
 $R = -9.0523197E-02$  SIGNIFICANCE OF R = SIGNIFICANT  $S_e = +6.2530927E-05$   
 $t = +2.3928376E+00$  SIGNIFICANCE OF t = SIGNIFICANT  $S_t = +4.6049121E-02$   
 $N = 695$  DEGREES OF FREEDOM = 693  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH





$F = +1.6484712E+02$  SIGNIFICANCE OF F = (-2.2459252E-01) \* X)  
 $R = -2.9317956E-01$  SIGNIFICANT  
 $t = +1.2839280E+01$  SIGNIFICANCE OF t = SIGNIFICANT  
 $N = 1755$  DEGREES OF FREEDOM = 1753  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH



STAGE 1, WING 2, IP-H1043, STRESS AT RUPTURE, LOW RATE, CHS-2.0

Figure 7

$Y = ((+1.1935827E+02) + (+4.2891356E-02) \times X)$   
 $F = +3.9107742E-01$  SIGNIFICANCE OF F = NOT SIGNIFICANT  $G_1 = +5.0486385E+01$   
 $R = +2.3748831E-02$  SIGNIFICANCE OF R = NOT SIGNIFICANT  $S_1 = +6.8586463E-02$   
 $I = +6.2536183E-01$  SIGNIFICANCE OF I = NOT SIGNIFICANT  $S_2 = +5.0508548E+01$   
 $N = 695$  DEGREES OF FREEDOM = 693  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

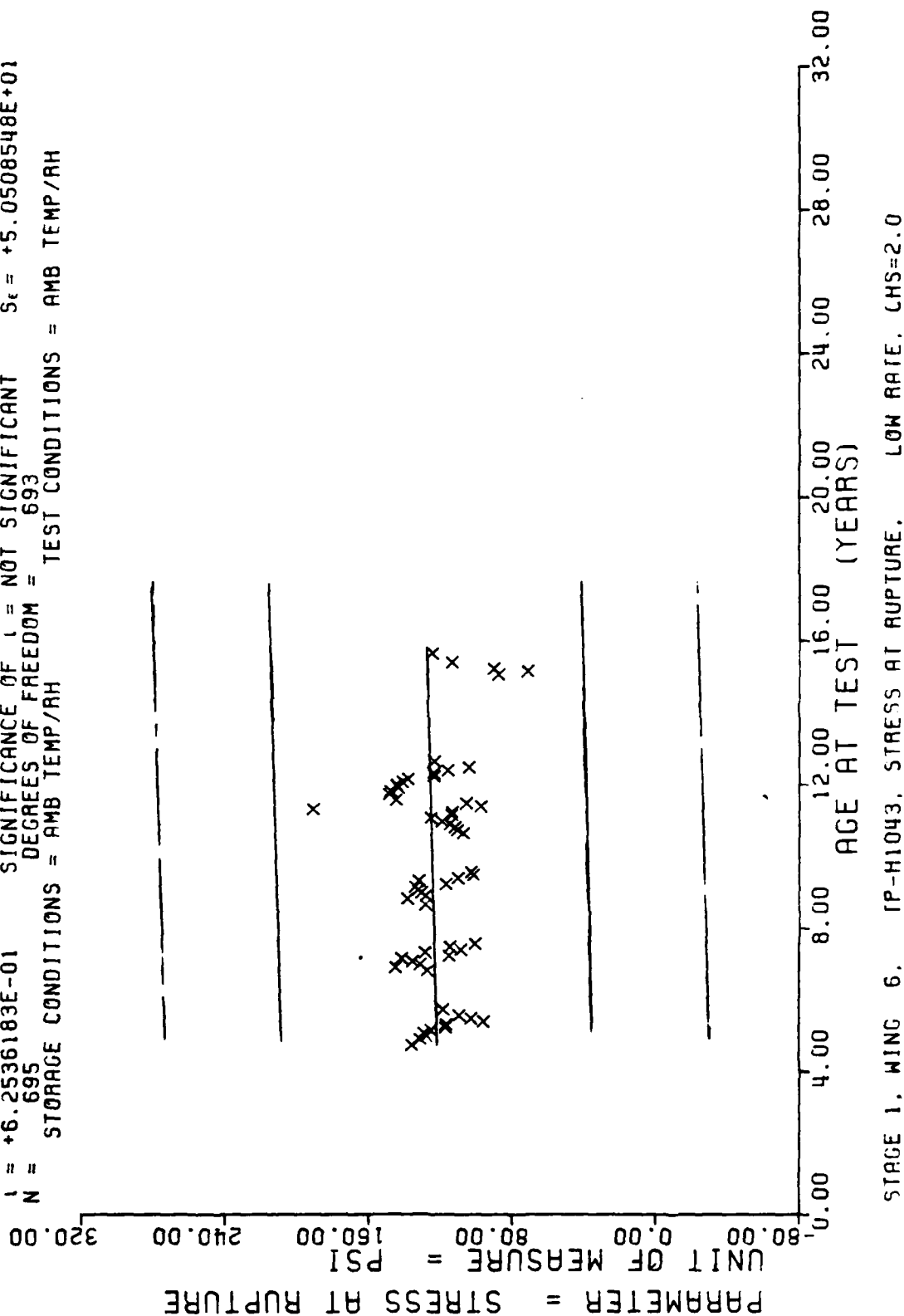


Figure 8

# ANALYSIS OF COVARIANCE COMPARISON OF REGRESSION LINES

```

ANALYSIS OF COVARIANCE • COMPARES REGRESSIONS • TP-H104E • CPHS=2.1 • STRS KURTUPE, W-2/6
*****
* * CORRECTED SUM SQUARES * REGRESSION * DEVIATIONS FROM REGRESSION
SOURCE * D.F.* SUM XX SUM YY SUM UY * U.F. * SUM SW. * MLAB SW.
*****
*ITHII. *
* * * * *
SMP2LP * 1754 * 0.29163E+07 -0.65499E+06 0.17114E+07 * -0.22489 * 1753 0.15643E+07 892.3R
SMPBLR * 694 * 0.54252E+06 25261. 0.17699E+07 * 0.42691E-01 * 693 0.17679E+07 2551.1
* * * * *
* * * * *
PCOLEP * 2440 * 0.54587E+07 -0.65173E+06 0.34044E+07 * -0.14265 * 2447 0.33650E+07 1375.1
* * * * *
* * * * *
DIFFERENCE BETWEEN SLOPES * 1 32717. 52717.
* * * * *
* * * * *
BFTWEN * 1 * 77156. 75536. 74032.
* * * * *
TOTAL * 2449 * 0.53557E+07 -0.55619E+06 0.55944E+07 * * 2448 0.34059E+07
* * * * *
* * * * *
BETWEEN ADJUSTED EARS * 1 0.1019.E+06 0.10192E+06
* * * * *

```

CONFIDENCE INTERVAL ESTIMATES :  $F = 2.00$  (DF = 633, 1755)

IF HOMOGENEITY OF RESIDUAL VARIANCES IS TRUE CONTINUE WITH STEP 2.

CO-PAID LAMIN OF RECESSION THEIR LINES : F= 24.02 (LF= 10.2446)

CO-PAID SOLUTION OF REGRESSION THEOREM :  $\Gamma = 24.02$  (cf. = 10.2446)

COMPASS REGRESSOR:  $F = 74.12$  (F = 10.447)



$Y = ((+2.8298920E+03) + (-3.7111880E+00) * X)$   
 $F = +7.4885681E+01$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = -2.0246208E-01$  SIGNIFICANCE OF R = SIGNIFICANT  
 $I = +8.6536513E+00$  SIGNIFICANCE OF I = SIGNIFICANT  
 $N = 1754$  DEGREES OF FREEDOM = 1752  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

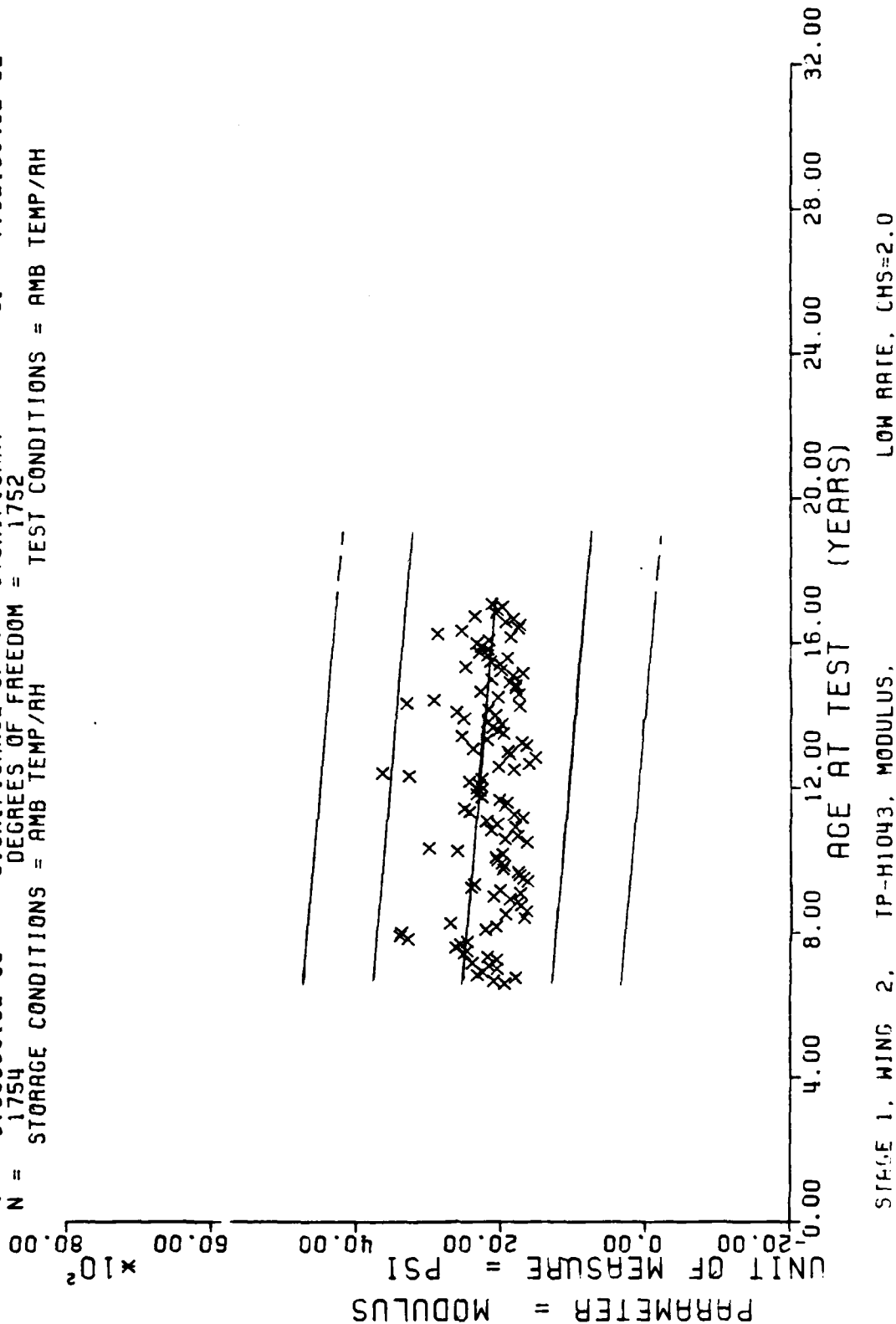


Figure 9

$Y = ((+2.2911536E+03) + (-1.7806905E-01) * X)$   
 F = +4.0031866E-01 SIGNIFICANCE OF F = NOT SIGNIFICANT  $\sigma_e = +2.0716877E+02$   
 R = -2.4027628E-02 SIGNIFICANCE OF R = NOT SIGNIFICANT  $S_e = +2.8143981E-01$   
 I = +6.3270741E-01 SIGNIFICANCE OF I = NOT SIGNIFICANT  $S_e = +2.0725834E+02$   
 N = 695 DEGREES OF FREEDOM = 693  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

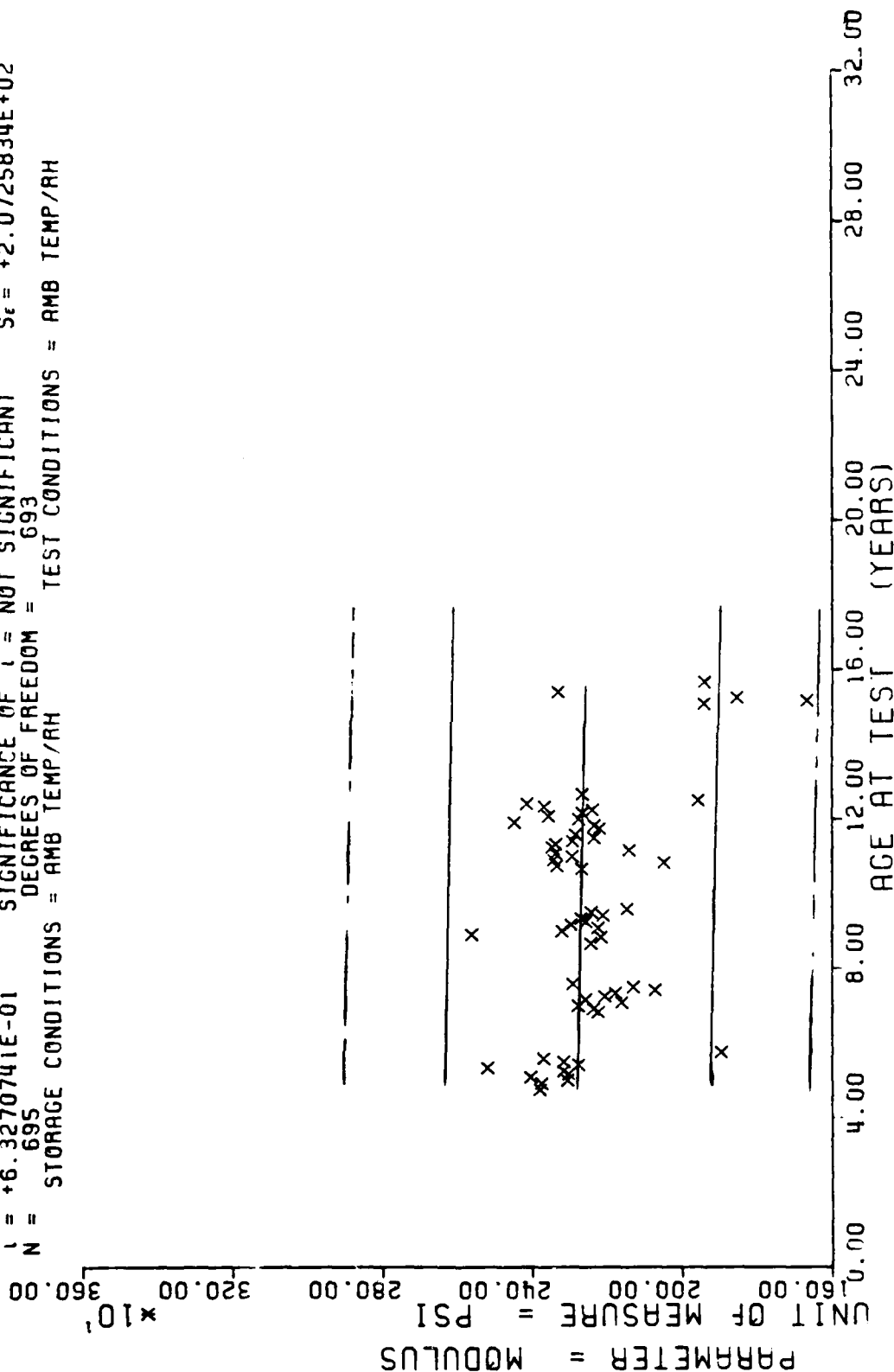


Figure 10

UNITED STATES DEPARTMENT OF JUSTICE  
FEDERAL BUREAU OF INVESTIGATION

.....

DEVIATIONS FROM REGULATION  
SUG. SW. 10E/11. SW.

[illegible]

25

DIFFERENCE BETWEEN SLICES

*	1	0.57678E+07	0.57678E+07
*			

$$0.0011 + 1.6 \times 10^{-7} = 1.60000011 \times 10^{-7}$$
$$\begin{aligned} r(TL) &= 26400 - 0.03244t + 0.7 - (-0.10444t + 0.8) - 0.10320t + 10 \\ &= 26400 - 0.03244t + 0.7 - 0.10444t + 0.8 - 0.10320t + 10 \\ &= 26400 - 0.23988t + 11.5 \end{aligned}$$

PLATE 1. 1145. 1146. 1147. 1148. 1149. 1150. 1151. 1152. 1153. 1154. 1155. 1156. 1157. 1158. 1159. 1160. 1161. 1162. 1163. 1164. 1165. 1166. 1167. 1168. 1169. 1170. 1171. 1172. 1173. 1174. 1175. 1176. 1177. 1178. 1179. 1180. 1181. 1182. 1183. 1184. 1185. 1186. 1187. 1188. 1189. 1190. 1191. 1192. 1193. 1194. 1195. 1196. 1197. 1198. 1199. 1200. 1201. 1202. 1203. 1204. 1205. 1206. 1207. 1208. 1209. 1210. 1211. 1212. 1213. 1214. 1215. 1216. 1217. 1218. 1219. 1220. 1221. 1222. 1223. 1224. 1225. 1226. 1227. 1228. 1229. 1230. 1231. 1232. 1233. 1234. 1235. 1236. 1237. 1238. 1239. 1240. 1241. 1242. 1243. 1244. 1245. 1246. 1247. 1248. 1249. 1250. 1251. 1252. 1253. 1254. 1255. 1256. 1257. 1258. 1259. 1260. 1261. 1262. 1263. 1264. 1265. 1266. 1267. 1268. 1269. 1270. 1271. 1272. 1273. 1274. 1275. 1276. 1277. 1278. 1279. 1280. 1281. 1282. 1283. 1284. 1285. 1286. 1287. 1288. 1289. 1290. 1291. 1292. 1293. 1294. 1295. 1296. 1297. 1298. 1299. 1300. 1301. 1302. 1303. 1304. 1305. 1306. 1307. 1308. 1309. 1310. 1311. 1312. 1313. 1314. 1315. 1316. 1317. 1318. 1319. 1320. 1321. 1322. 1323. 1324. 1325. 1326. 1327. 1328. 1329. 1330. 1331. 1332. 1333. 1334. 1335. 1336. 1337. 1338. 1339. 1340. 1341. 1342. 1343. 1344. 1345. 1346. 1347. 1348. 1349. 1350. 1351. 1352. 1353. 1354. 1355. 1356. 1357. 1358. 1359. 1360. 1361. 1362. 1363. 1364. 1365. 1366. 1367. 1368. 1369. 1370. 1371. 1372. 1373. 1374. 1375. 1376. 1377. 1378. 1379. 1380. 1381. 1382. 1383. 1384. 1385. 1386. 1387. 1388. 1389. 1390. 1391. 1392. 1393. 1394. 1395. 1396. 1397. 1398. 1399. 1400. 1401. 1402. 1403. 1404. 1405. 1406. 1407. 1408. 1409. 1410. 1411. 1412. 1413. 1414. 1415. 1416. 1417. 1418. 1419. 1420. 1421. 1422. 1423. 1424. 1425. 1426. 1427. 1428. 1429. 1430. 1431. 1432. 1433. 1434. 1435. 1436. 1437. 1438. 1439. 1440. 1441. 1442. 1443. 1444. 1445. 1446. 1447. 1448. 1449. 1450. 1451. 1452. 1453. 1454. 1455. 1456. 1457. 1458. 1459. 1460. 1461. 1462. 1463. 1464. 1465. 1466. 1467. 1468. 1469. 1470. 1471. 1472. 1473. 1474. 1475. 1476. 1477. 1478. 1479. 1480. 1481. 1482. 1483. 1484. 1485. 1486. 1487. 1488. 1489. 1490. 1491. 1492. 1493. 1494. 1495. 1496. 1497. 1498. 1499. 1500. 1501. 1502. 1503. 1504. 1505. 1506. 1507. 1508. 1509. 1510. 1511. 1512. 1513. 1514. 1515. 1516. 1517. 1518. 1519. 1520. 1521. 1522. 1523. 1524. 1525. 1526. 1527. 1528. 1529. 1530. 1531. 1532. 1533. 1534. 1535. 1536. 1537. 1538. 1539. 1540. 1541. 1542. 1543. 1544. 1545. 1546. 1547. 1548. 1549. 1550. 1551. 1552. 1553. 1554. 1555. 1556. 1557. 1558. 1559. 1560. 1561. 1562. 1563. 1564. 1565. 1566. 1567. 1568. 1569. 1570. 1571. 1572. 1573. 1574. 1575. 1576. 1577. 1578. 1579. 1580. 1581. 1582. 1583. 1584. 1585. 1586. 1587. 1588. 1589. 1590. 1591. 1592. 1593. 1594. 1595. 1596. 1597. 1598. 1599. 1600. 1601. 1602. 1603. 1604. 1605. 1606. 1607. 1608. 1609. 1610. 1611. 1612. 1613. 1614. 1615. 1616. 1617. 1618. 1619. 1620. 1621. 1622. 1623. 1624. 1625. 1626. 1627. 1628. 1629. 1630. 1631. 1632. 1633. 1634. 1635. 1636. 1637. 1638. 1639. 1640. 1641. 1642. 1643. 1644. 1645. 1646. 1647. 1648. 1649. 1650. 1651. 1652. 1653. 1654. 1655. 1656. 1657. 1658. 1659. 1660. 1661. 1662. 1663. 1664. 1665. 1666. 1667. 1668. 1669. 1670. 1671. 1672. 1673. 1674. 1675. 1676. 1677. 1678. 1679. 1680. 1681. 1682. 1683. 1684. 1685. 1686. 1687. 1688. 1689. 1690. 1691. 1692. 1693. 1694. 1695. 1696. 1697. 1698. 1699. 1700. 1701. 1702. 1703. 1704. 1705. 1706. 1707. 1708. 1709. 1710. 1711. 1712. 1713. 1714. 1715. 1716. 1717. 1718. 1719. 1720. 1721. 1722. 1723. 1724. 1725. 1726. 1727. 1728. 1729. 1730. 1731. 1732. 1733. 1734. 1735. 1736. 1737. 1738. 1739. 1740. 1741. 1742. 1743. 1744. 1745. 1746. 1747. 1748. 1749. 1750. 1751. 1752. 1753. 1754. 1755. 1756. 1757. 1758. 1759. 1760. 1761. 1762. 1763. 1764. 1765. 1766. 1767. 1768. 1769. 1770. 1771. 1772. 1773. 1774. 1775. 1776. 1777. 1778. 1779. 1780. 1781. 1782. 1783. 1784. 1785. 1786. 1787. 1788. 1789. 1790. 1791. 1792. 1793. 1794. 1795. 1796. 1797. 1798. 1799. 1800. 1801. 1802. 1803. 1804. 1805. 1806. 1807. 1808. 1809. 1810. 1811. 1812. 1813. 1814. 1815. 1816. 1817. 1818. 1819. 1820. 1821. 1822. 1823. 1824. 1825. 18

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[illegible]

STAGE 1, WHICH 2, TD-H1043, STRAIN AT MAX STRESS, HIGH FATI, CUS=1750.0

This sample size summary is applicable to figures 11, 13, 15, 17, and 19

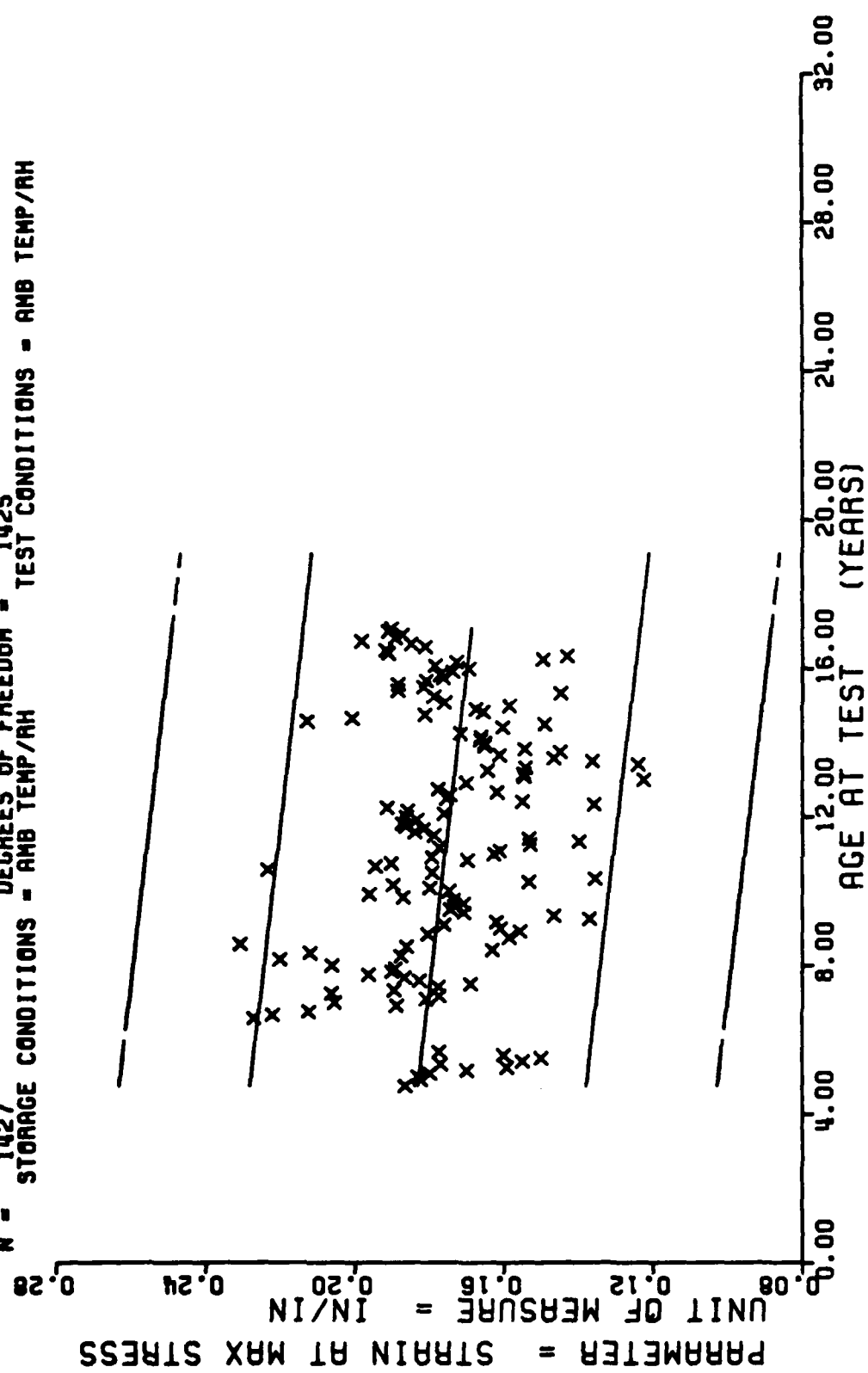
\*\*\* SA HILL SITE SUMMARY \*\*\*

AGE (LBS)	NO SAMP	AGE (MO)	NO SAMP
82	3	132	22
83	15	133	26
84	24	134	17
85	24	135	15
86	18	136	25
87	21	137	21
88	19	138	18
89	20	139	6
90	19	140	3
91	9	141	6
105	3	142	15
107	3	143	27
108	9	144	27
109	23	145	21
110	20	146	24
111	21	147	18
112	21	148	15
113	21	149	15
114	10	150	12
115	22		
116	12		
127	3		
129	3		
130	9		
131	21		

STAGE 1. VING 0. TP-H1043. STEADY AT MAX STRESS. HIGH RATE. CHS=1750.0

This sample size summary is applicable to figures 12, 14, 16, 18 and 20

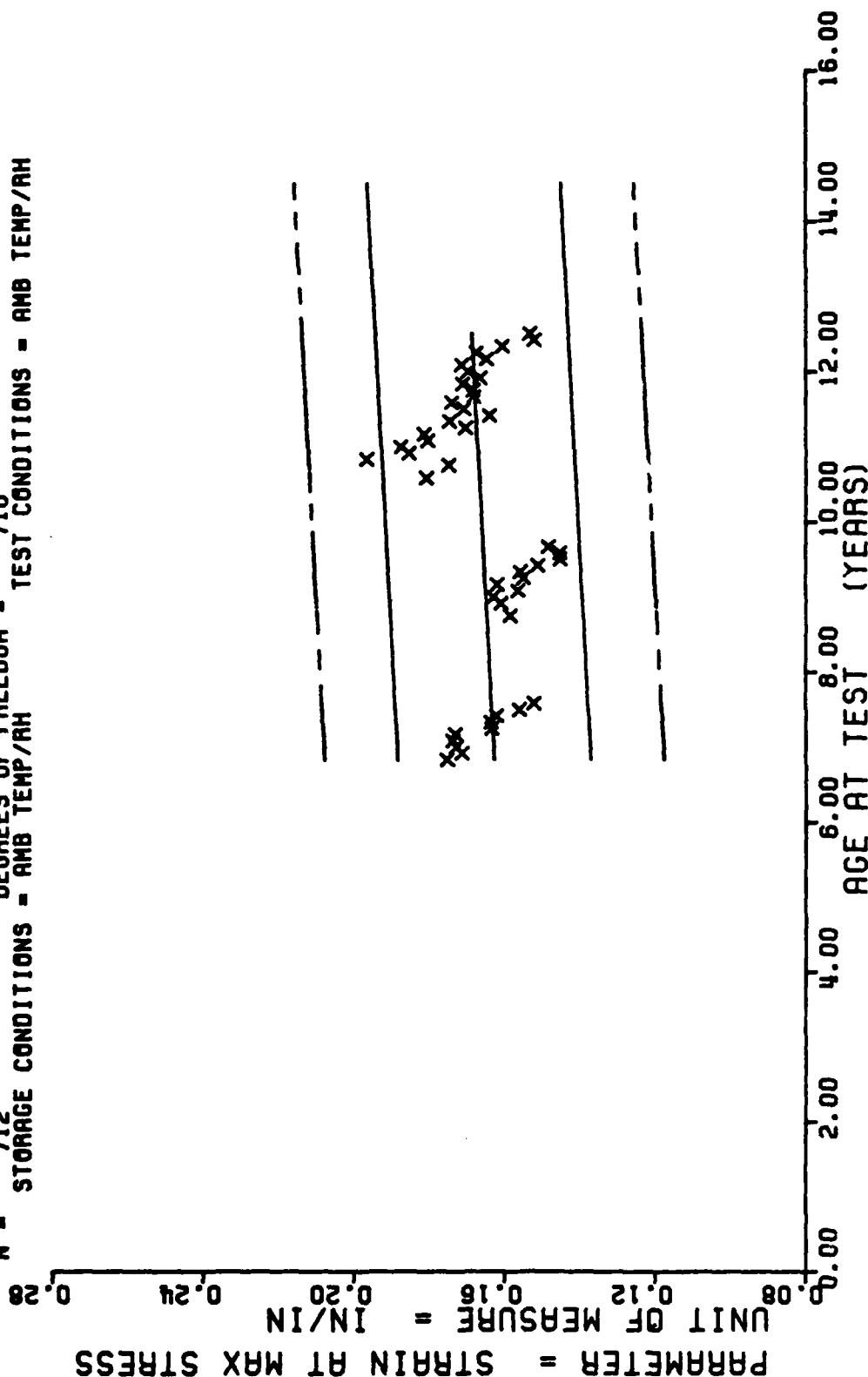
$F = +3.2622844E+01$  SIGNIFICANCE OF F = SIGNIFICANT  $\alpha = +2.7043880E-02$   
 $R = -1.5004990E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_e = +1.7167754E-05$   
 $t = +5.7291225E+00$  SIGNIFICANCE OF t = SIGNIFICANT  $S_e = +2.6747061E-02$   
 $N = 1427$  DEGREES OF FREEDOM = 1425  
 STORAGE CONDITIONS = AMB TEMP/AM TEST CONDITIONS = AMB TEMP/AM



STAGE 1. WING 2. TP-H1043. STRAIN AT MAX STRESS. HIGH RATE. CHS=1750.0

Figure 11

$Y = (( +1.5541790E-01 ) + ( +8.7935927E-05 ) \times X)$   
 $F = +1.2422184E+01$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = +1.3113031E-01$  SIGNIFICANCE OF R = SIGNIFICANT  
 $t = +3.5245090E+00$  SIGNIFICANCE OF t = SIGNIFICANT  
 $N = 712$  DEGREES OF FREEDOM = 710  
 STORAGE CONDITIONS = AMB TEMP/AM TEST CONDITIONS = AMB TEMP/AM



STAGE 1, WING 6. TP-H1043. STRAIN AT MAX STRESS. HIGH RATE. CHS=1750.0

Figure 12

THE UNIVERSITY OF MICHIGAN

[illegible]

		* 2136	1.1791	U.S. 47E-08
POLICE # 2127#	0.785E+07	-0.791E-04*	1.1501	U.S. 47E-01

CPUFE	=	1*	0.621167+07	-7/99A	+0.621167-01 *
TOTAL	=	1.1798	0.621067+07	-11.000	+2357 1.02677

一一一

[illegible]

1112

$(\text{COP})_{\text{eff}} = 117.4, \eta = 1 : 0.0017, \text{COP} = 117.4$   
 $(\text{COP})_{\text{eff}} = 117.4, \eta = 1 : 0.0017, \text{COP} = 117.4$



$Y = ((+4.9569479E+02) + (-1.2070152E-01) * X)$   
 $F = +1.8124830E+01$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma_1 = +4.4435547E+01$   
 $R = -1.1206888E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_0 = +2.8351481E-02$   
 $t = +4.2573267E+00$  SIGNIFICANCE OF t = SIGNIFICANT  $S_1 = +4.4171114E+01$   
 $N = 1427$  DEGREES OF FREEDOM = 1425  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

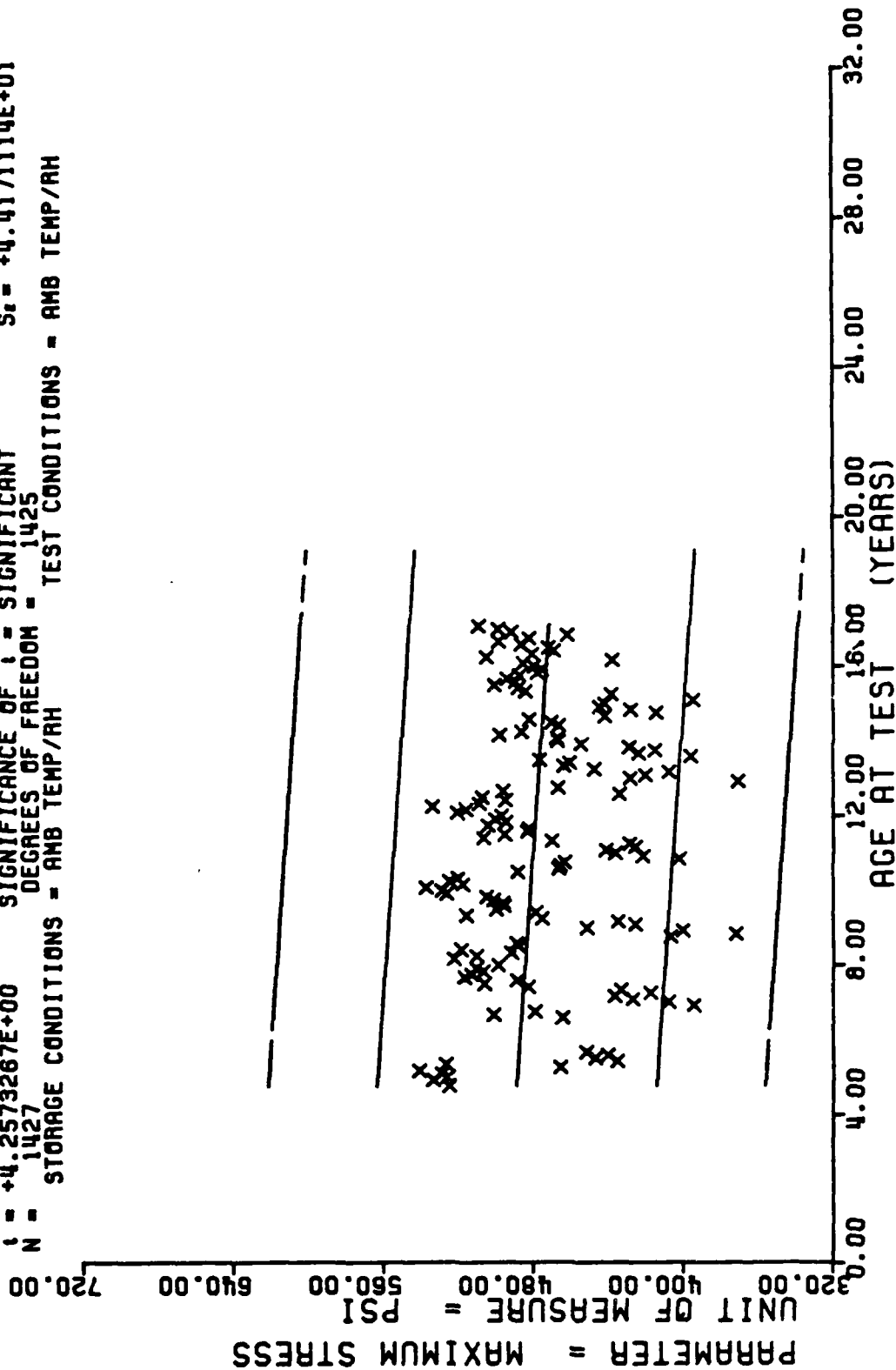
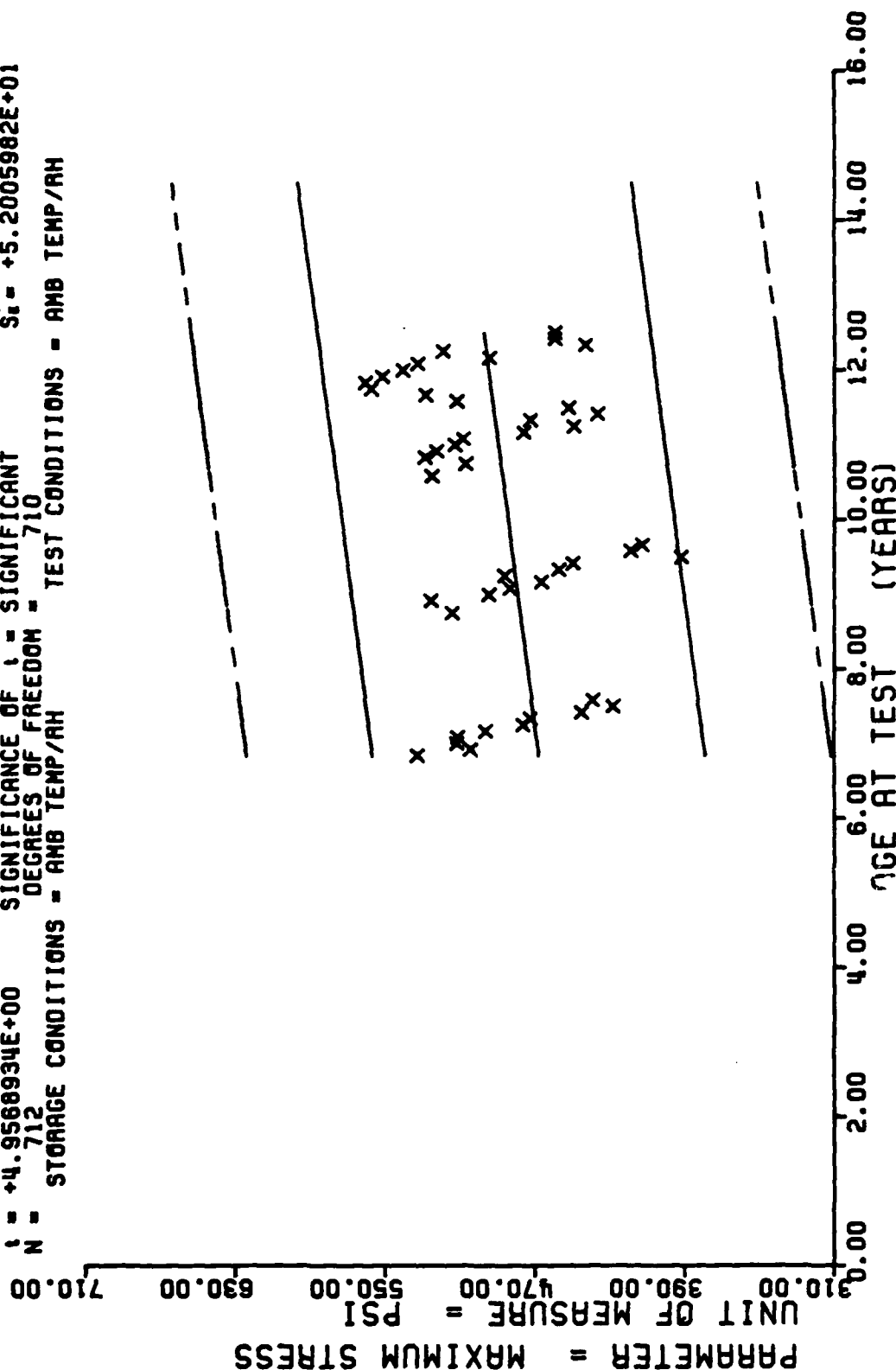


Figure 13

$Y = ((+4.3251480E+02) + (+4.2895280E-01) * X)$   
 $F = +2.4570792E+01$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma_r = +5.2860994E+01$   
 $R = +1.8289117E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_e = +8.6415576E-02$   
 $t = +4.9568934E+00$  SIGNIFICANCE OF t = SIGNIFICANT  $S_t = +5.2005982E+01$   
 $N = 712$  DEGREES OF FREEDOM = 710  
 STORAGE CONDITIONS = AMB TEMP/AH TEST CONDITIONS = AMB TEMP/AH

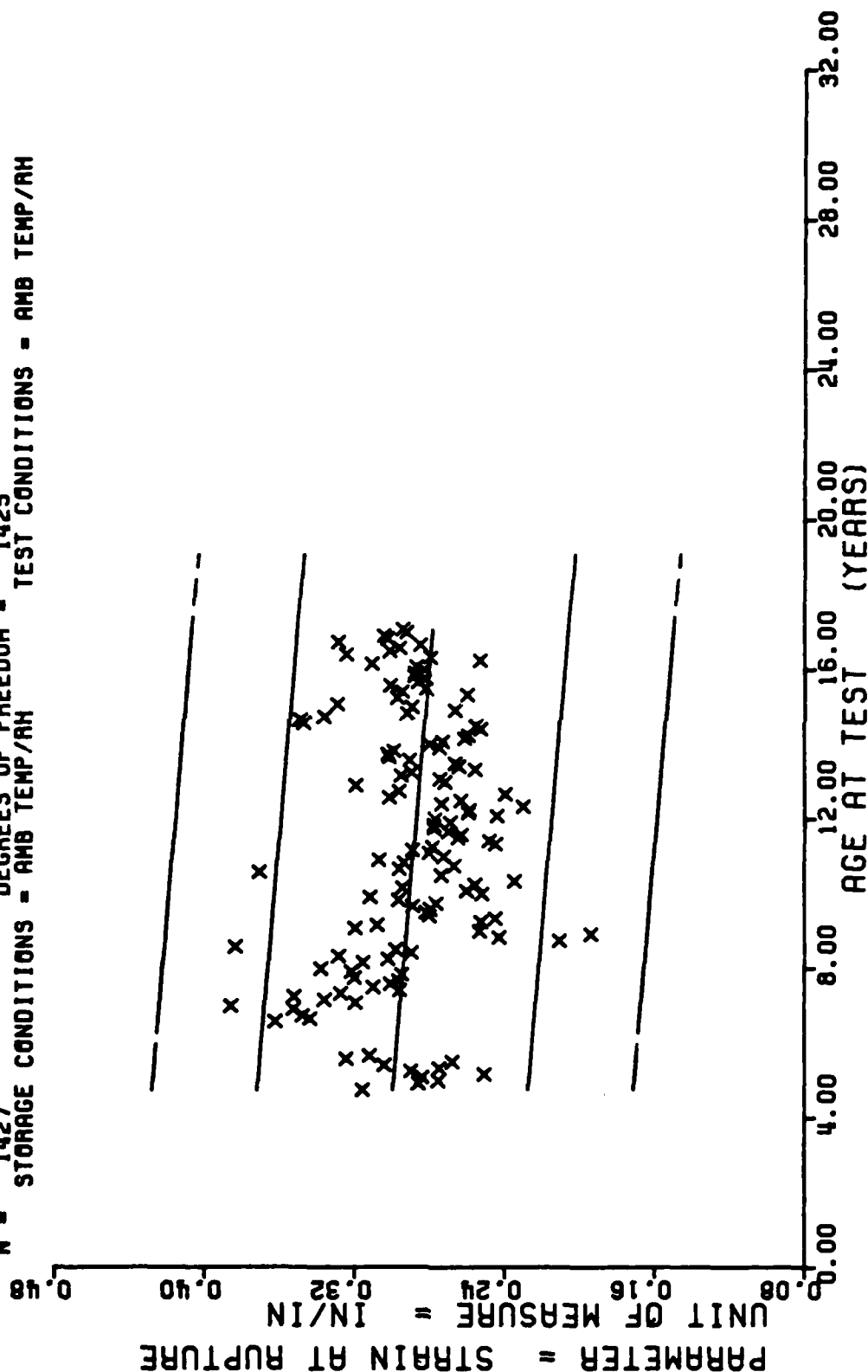


STAGE 1, WING 6. TP-H1043. MAXIMUM STRESS. HIGH RATE. CHS-1750.0

Figure 14



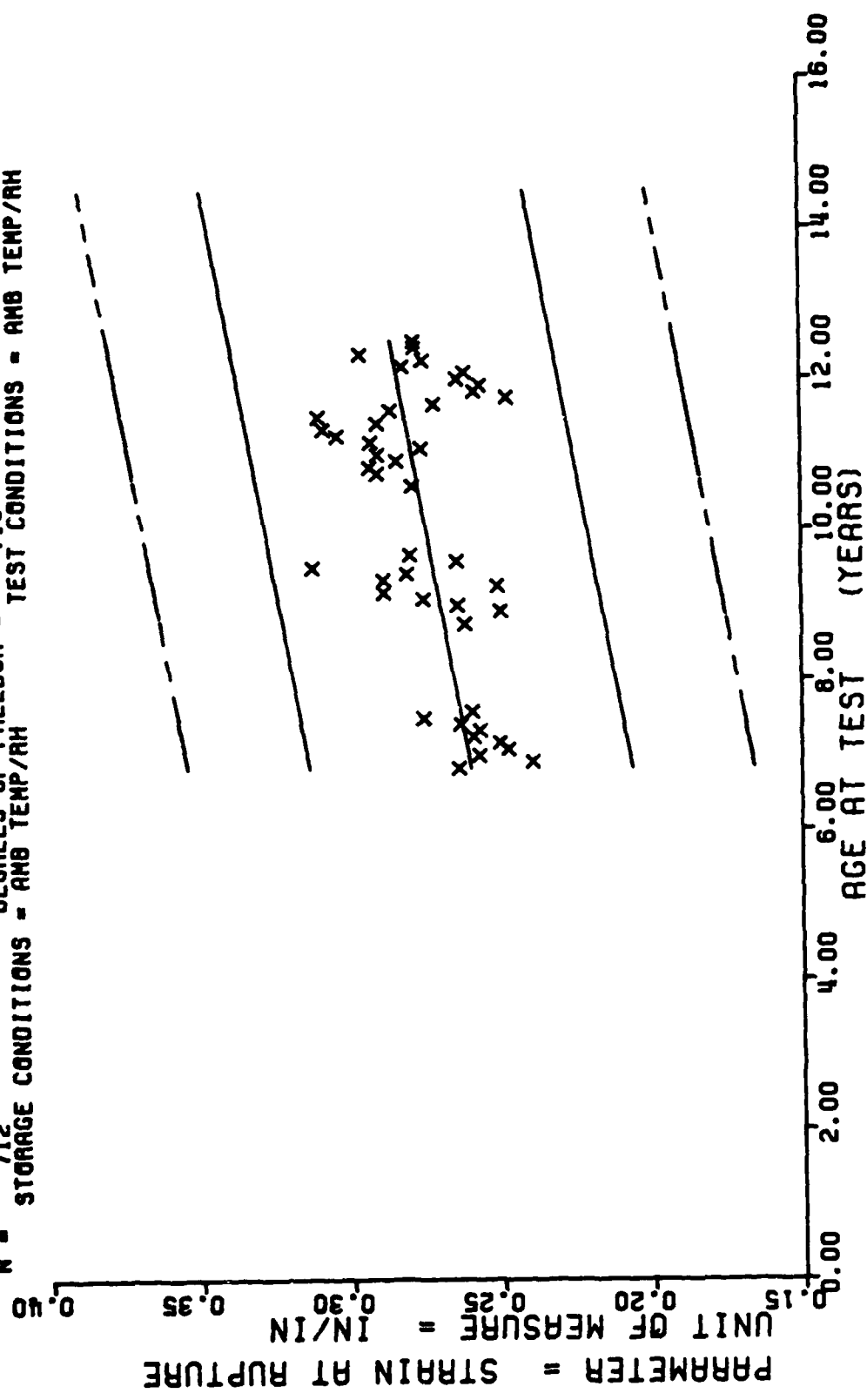
$Y = ((+3.0778857E-01) + (-1.4608721E-04) * X)$   
 $F = +2.8331608E+01$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma_1 = +4.3167986E-02$   
 $R = -1.3962180E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_1 = +2.7445844E-05$   
 $t = +5.3227444E+00$  SIGNIFICANCE OF t = SIGNIFICANT  $S_2 = +4.2760147E-02$   
 $N = 1427$  DEGREES OF FREEDOM = 1425  
 STORAGE CONDITIONS = AMB TEMP/AM TEST CONDITIONS = AMB TEMP/AM



STAGE 1, WING 2, TP-H1043, STRAIN AT RUPTURE, HIGH RATE, CHS=1750.0

Figure 15

$Y = ((+2.2861467E-01) + (+3.8023994E-04) \times X)$   
 $F = +5.3989088E+01$  SIGNIFICANCE OF F = SIGNIFICANT  $G = +3.2451213E-02$   
 $R = +2.6445591E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_0 = +5.2039305E-05$   
 $I = +7.3067837E+00$  SIGNIFICANCE OF I = SIGNIFICANT  $S_1 = +3.1317909E-02$   
 $N = 712$  DEGREES OF FREEDOM = 710  
 STORAGE CONDITIONS = AMB TEMP/AM TEST CONDITIONS = AMB TEMP/AM

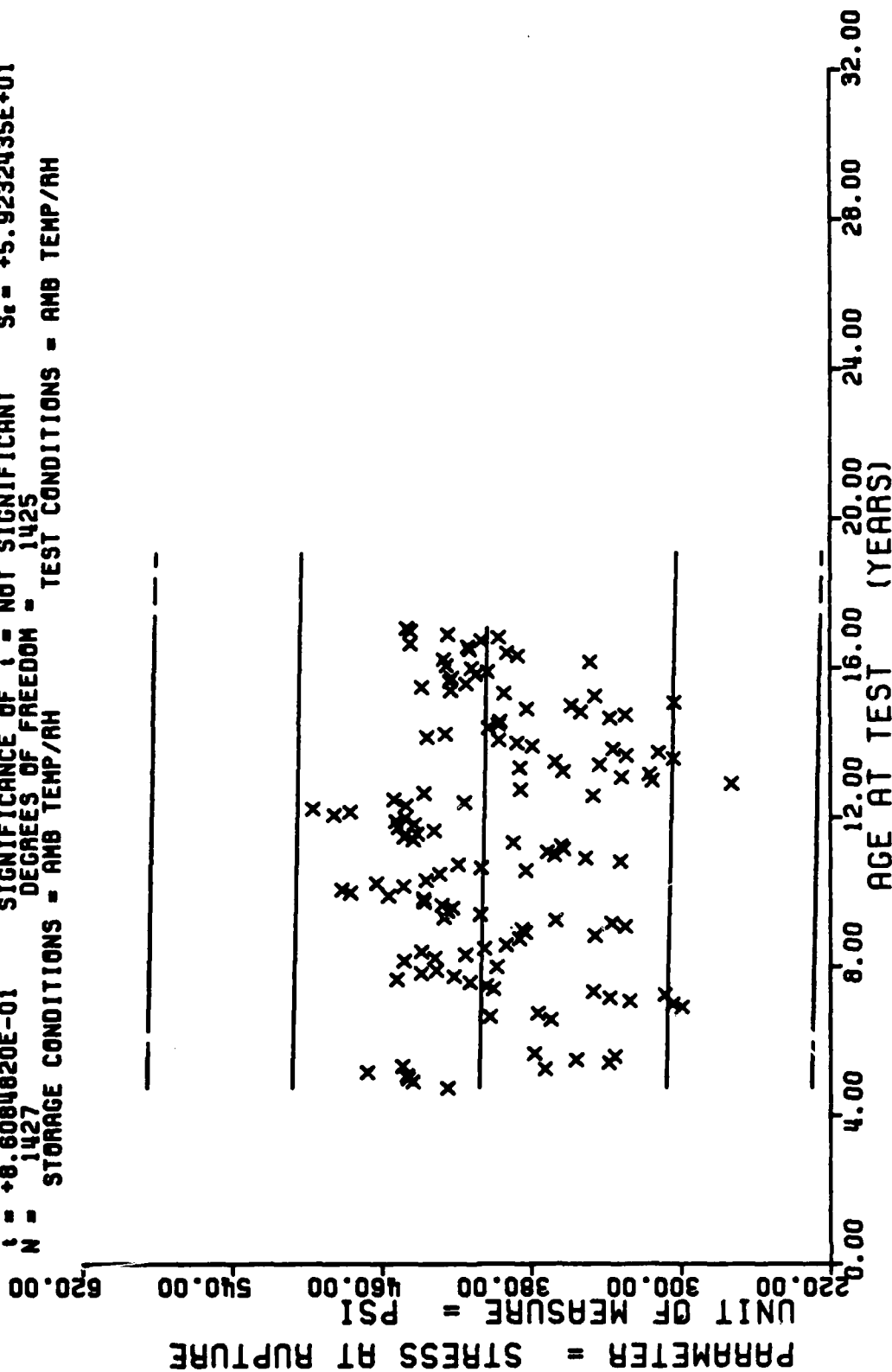


STAGE 1. WING 6. TP-H1043. STRAIN AT RUPTURE. HIGH RATE, CHS=1750.0

Figure 16



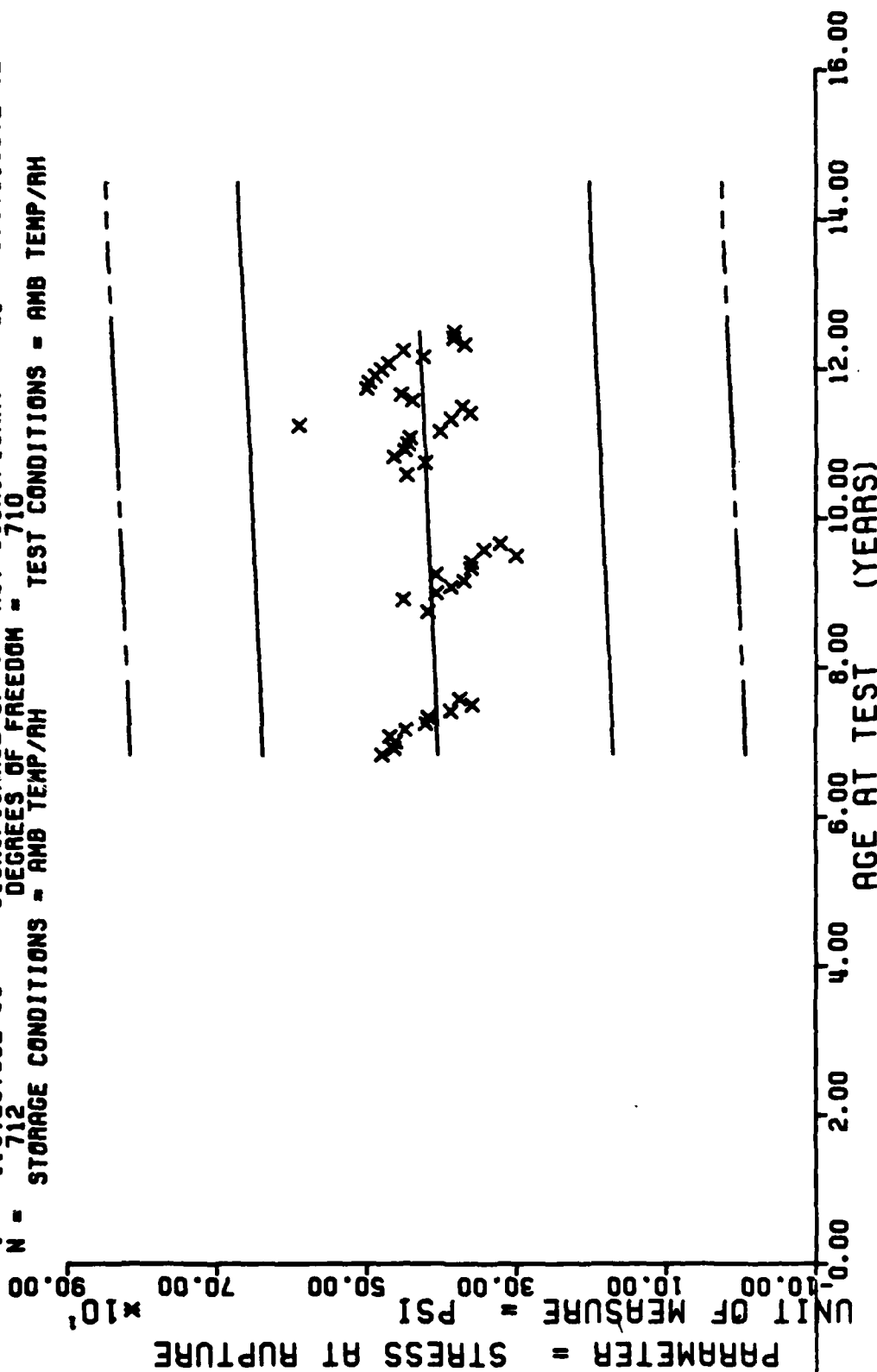
$Y = ((+4.0950908E+02) + (-3.2726309E-02) * X)$   
 $F = +7.4105963E-01$  SIGNIFICANCE OF F = NOT SIGNIFICANT  $S_1 = +5.9227057E+01$   
 $R = -2.2798498E-02$  SIGNIFICANCE OF R = NOT SIGNIFICANT  $S_2 = +3.8018676E-02$   
 $t = +8.6084620E-01$  SIGNIFICANCE OF t = NOT SIGNIFICANT  $S_3 = +5.9232435E+01$   
 $N = 1427$  DEGREES OF FREEDOM = 1425  
 STORAGE CONDITIONS = AMB TEMP/AH TEST CONDITIONS = AMB TEMP/AH



STAGE 1. WING 2. TP-H1043. STRESS AT RUPTURE. HIGH RATE. CHS=1750.0

Figure 17

$Y = ((+3.7477006E+02) + (+3.5808746E-01) \times X)$   
 $F = +2.4739415E+00$  SIGNIFICANCE OF F = NOT SIGNIFICANT  $\sigma^2 = +1.3715301E+02$   
 $R = +5.8926442E-02$  SIGNIFICANCE OF R = NOT SIGNIFICANT  $S_e = +2.2766402E-01$   
 $t = +1.5728768E+00$  SIGNIFICANCE OF t = NOT SIGNIFICANT  $S_e = +1.3701107E+02$   
 $N = 712$  DEGREES OF FREEDOM = 710  
 STORAGE CONDITIONS = AMB TEMP/AH TEST CONDITIONS = AMB TEMP/AH



STAGE 1. WING 6. TP-H1043, STRESS AT RUPTURE, HIGH RATE, CHS-1750.0

Figure 18



## ANALYSIS OF COVARIANCE COMPARISONS OF REGRESSION LINES

ANALYSIS OF COVARIATES REGRESSIONS), TP=111043, C.F.S.=1750, STPS=100106, W=270

SOURCE	U.F.	SUM XX	CORRELATED SUM SQUARES	SUM XY	SUM YY	* REGRESSION *	* COEFFICIENTS *	U.F.	SUM SQ.	DEVIATIONS FROM MEAN SS.	MEAN SQ.
SRP2PB	1426	0.29273E+07	-79441.	0.50022E+07	0.50022E+07	-0.42726E-01	1425	0.49999E+07	3501.5		
SRP610	711	0.36213E+06	6.12969E+06	0.13755E+06	0.13755E+06	0.45800E	710	0.13529E+06	18772.		
PCOUFI	2137	0.27895E+07	51251.	0.10777E+06	0.10777E+06	0.1015E-01	2136	0.10571E+06	6602.9		
DIFFERENCE BETWEEN SLOPES							1	48136.	48136.		
DIFFERENCE	1	0.21117E+06	-0.12931E+06	75072.							
TOTAL	2138	0.60066E+07	-75063.	0.19456E+06			2137	0.18454E+06			
ADJUSTED MEANS							1	75012.	75012.		

COMPARISON BETWEEN FESTIVAL VARIABLES:  $F = 5.55 (df = 7) (0.1425)$

• 24715 HITA 30001000 700L ST 3000H1000A 70001000 17 11111111111111 411

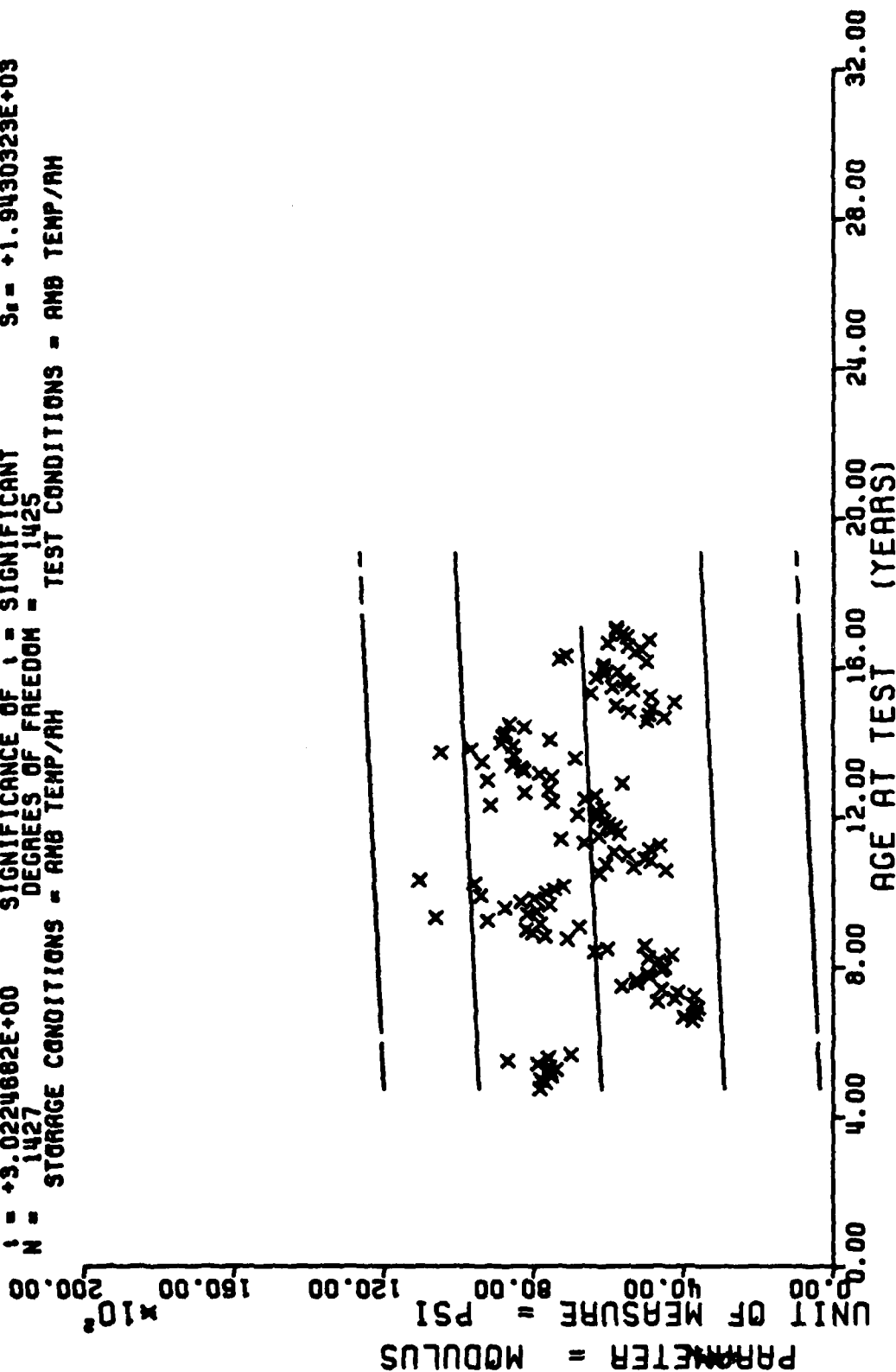
195701 5.115 = 1 : 57678 71771 0075737331 10 0007404000

[illegible]

## STATISTICALLY DIFF.

**STATISTICALLY DIFF.**  
**STATISTICALLY DIFF.**

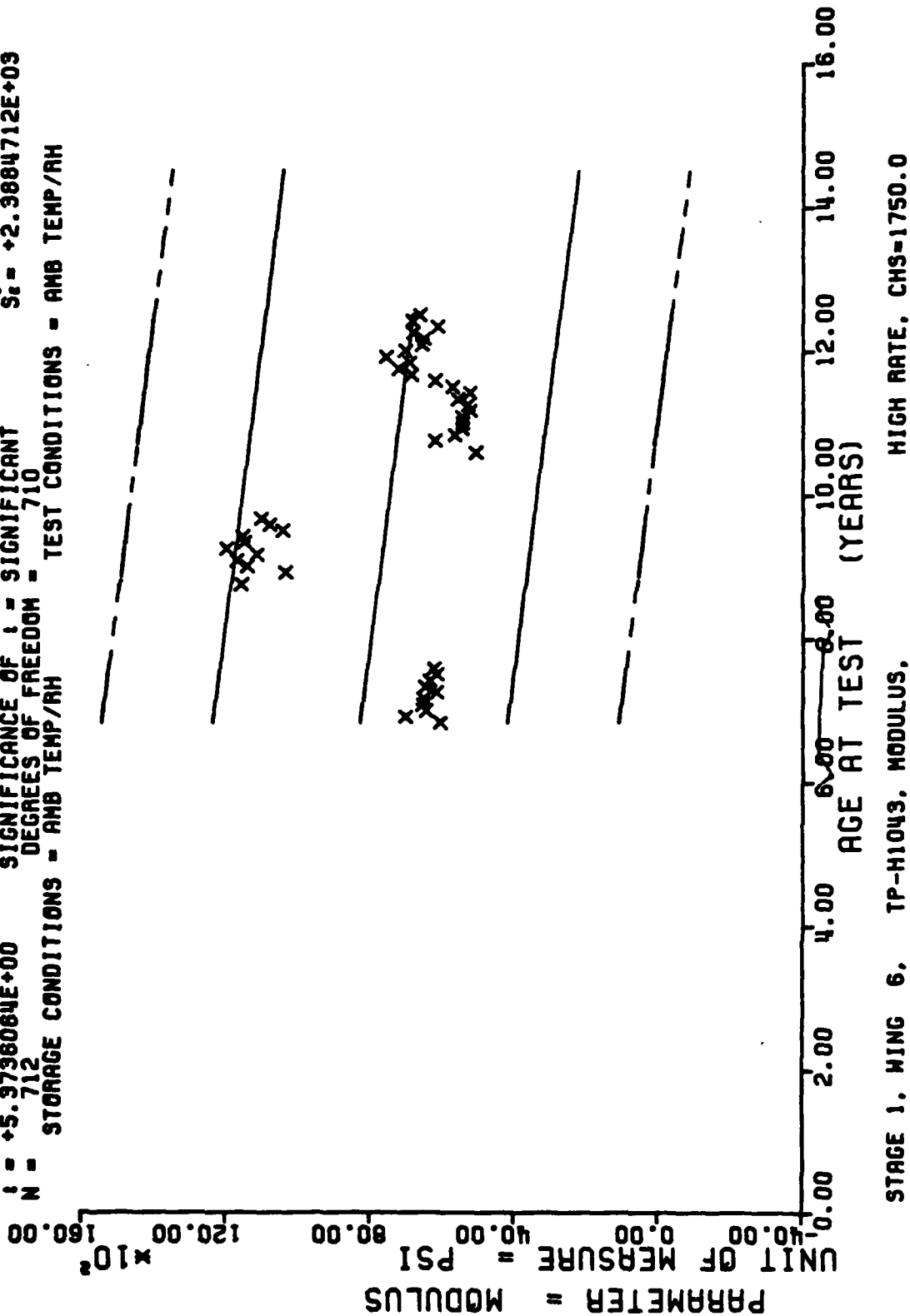
$Y = ((+5.9570324E+03) + (+3.7894605E+00) \times X)$   
 $F = +9.1353145E+00$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = +7.9811723E-02$  SIGNIFICANCE OF R = SIGNIFICANT  
 $S = +3.0224662E+00$  SIGNIFICANCE OF S = SIGNIFICANT  
 $N = 1427$  DEGREES OF FREEDOM = 1425  
 STORAGE CONDITIONS = AMB TEMP/AH TEST CONDITIONS = AMB TEMP/AH



STAGE 1, WING 2. TP-H1043, MODULUS. HIGH RATE, CHS-1750.0

Figure 19

$Y = ((+1.0009277E+04) + (-2.1326746E+01) * X)$   
 $F = +2.8875846E+01$  SIGNIFICANCE OF F = SIGNIFICANT  $G = +2.4348426E+03$   
 $R = -1.9768794E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_0 = +3.9687957E+00$   
 $I = +5.9736084E+00$  SIGNIFICANCE OF I = SIGNIFICANT  $S_1 = +2.3884712E+03$   
 $N = 712$  DEGREES OF FREEDOM = 710  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH



ANALYSIS OF COMPARISON  
OF THE TWO GROUPS OF  
STUDENTS IN SATISFACTION

[illegible]

```

* 1* 0.2111E+07 -0.2501E+07 0.4749E+09 *
* 213* 0.5100E+07 -0.6156E+07 0.1007E+11 *
* 2137 0.1004E+11

```

DATE	DESCRIPTION	AMOUNT	BALANCE
1994-01-01	OPENING BALANCE	100.00	100.00
1994-01-15	PAYROLL	10.00	90.00
1994-01-31	RENT	20.00	70.00
1994-02-15	UTILITIES	5.00	65.00
1994-02-28	CLOSING BALANCE		65.00

STATISTICALLY DIFF.  
STATISTICALLY DIFF.

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
56	5	92	9	120	20	148	15	185	20		
58	5	93	6	121	40	149	9	186	20		
59	10	94	11	123	20	150	5	187	10		
60	5	95	6	124	13	151	5	190	10		
61	15	96	3	126	6	152	5	191	10		
62	10	98	1	127	3	153	5	192	10		
63	10	99	1	128	12	155	5	193	50		
64	10	101	10	130	6	156	10	194	10		
65	10	102	11	131	15	158	25	195	35		
66	10	103	15	132	19	159	5	196	25		
67	11	105	10	133	25	160	5	197	10		
68	5	106	5	134	14	161	15	198	10		
78	2	107	15	135	17	163	5	199	5		
80	4	108	20	136	11	164	5	200	15		
81	1	109	15	137	21	165	10	201	5		
82	2	110	15	138	3	166	25	202	15		
83	1	111	10	139	9	167	10	203	10		
84	1	112	20	140	15	168	15	204	20		
85	5	113	15	141	9	169	5	205	15		
86	4	114	20	142	18	170	10				
87	3	115	10	143	12	171	15				
88	1	116	20	144	8	173	15				
89	7	117	15	145	16	174	5				
90	5	118	40	146	18	181	10				
91	4	119	10	147	3	184	5				

STAGE 1, WING 2. TP-H1043. HARDNESS SHORE A, INITIAL

This sample size summary is applicable to figures 21 and 23

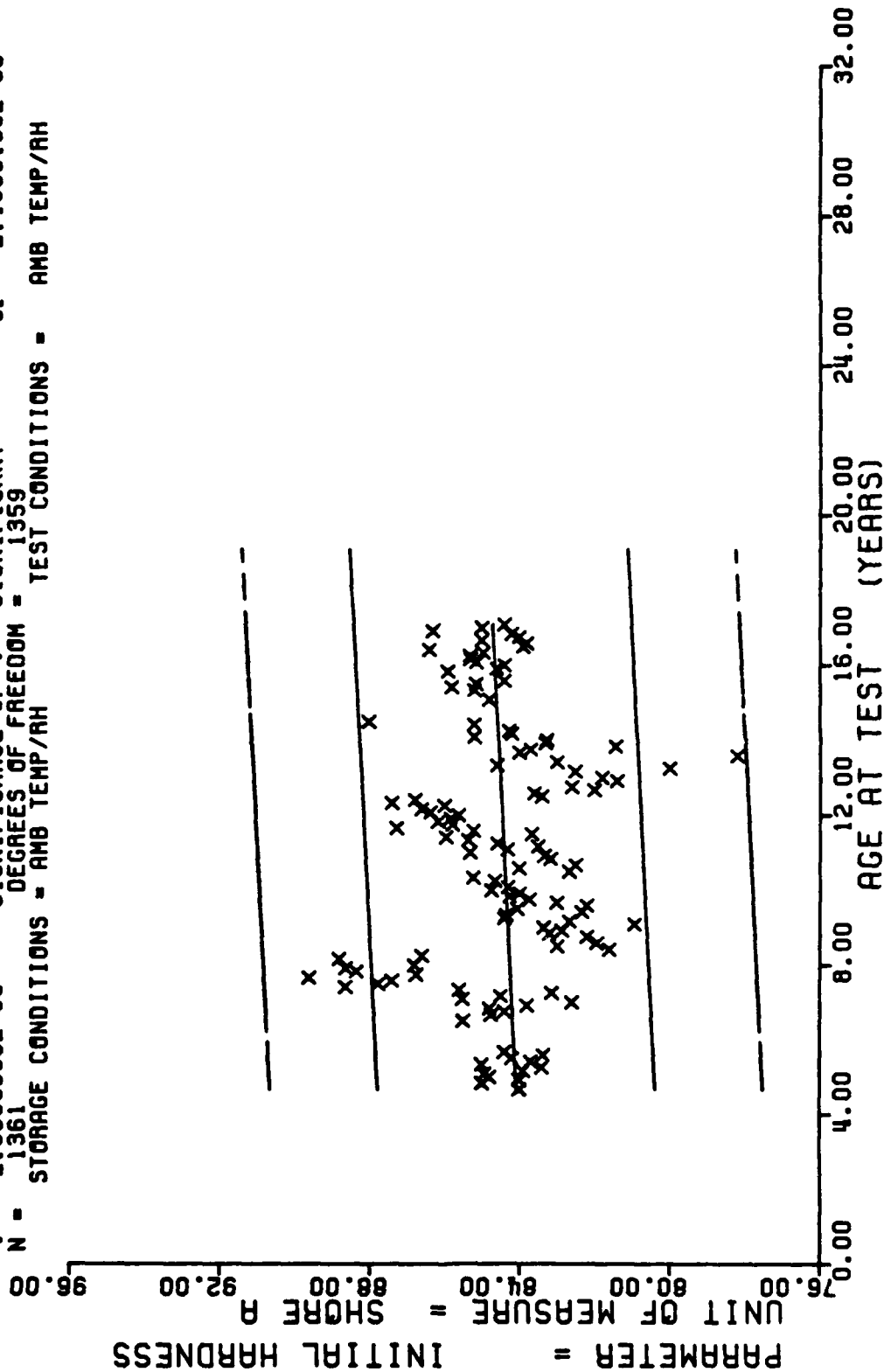
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
82	12	146	35
83	15	147	40
84	24		
85	27		
86	15		
87	18		
88	21		
89	21		
90	9		
91	9		
106	5		
107	5		
108	10		
109	20		
110	25		
111	15		
112	15		
113	20		
114	15		
115	20		
141	15		
142	15		
143	30		
144	55		
145	25		

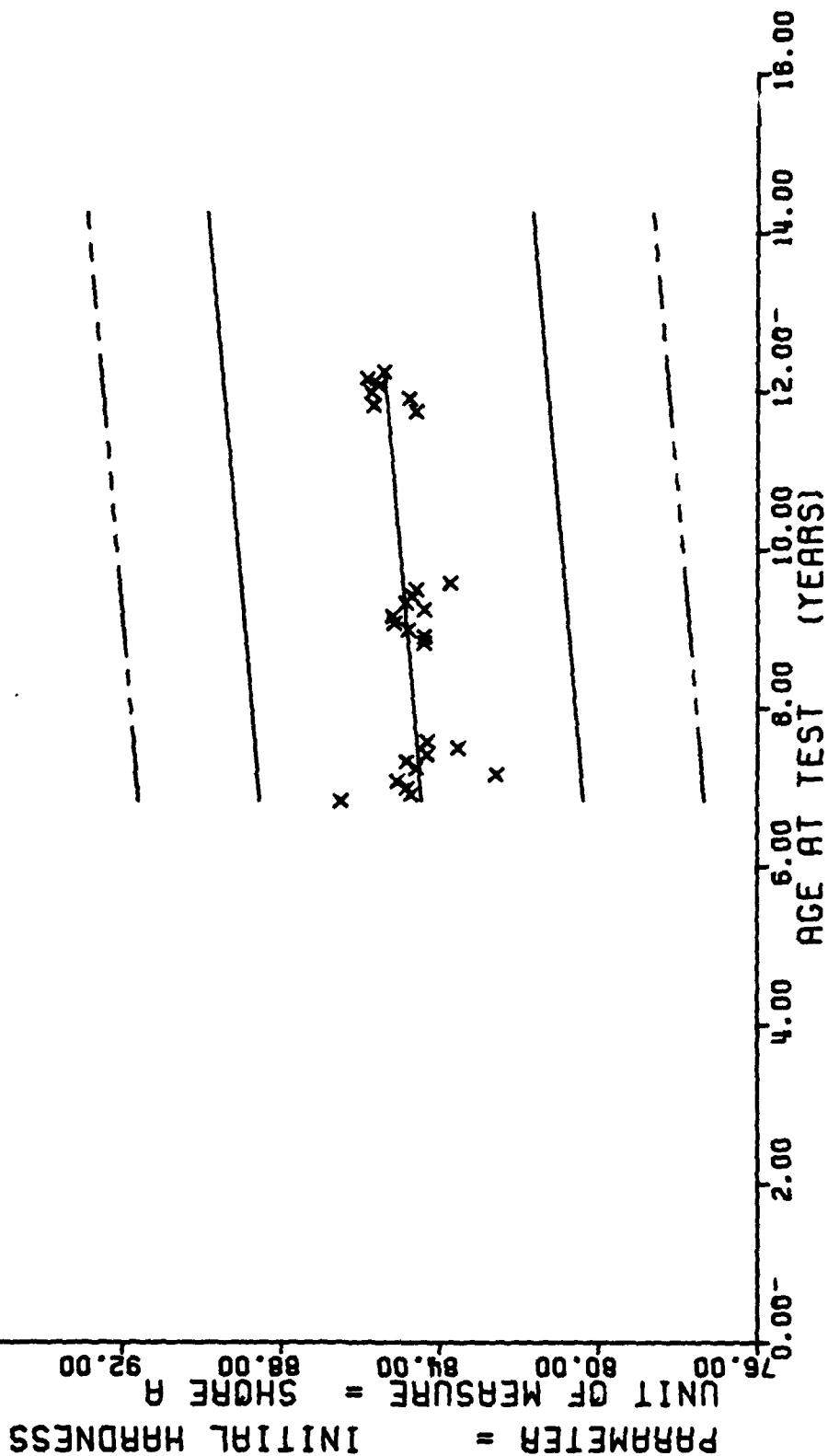
STAGE 1, WING 6, TP-H1043, HARDNESS SHORE A, INITIAL

This sample size summary is applicable to figures 22 and 24

$Y = ((+8.3844469E+01) + (+4.2421133E-03) \times X)$   
 $F = +8.2137571E+00$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = +7.7509104E-02$  SIGNIFICANCE OF R = SIGNIFICANT  
 $t = +2.8659653E+00$  SIGNIFICANCE OF t = SIGNIFICANT  
 $N = 1361$  DEGREES OF FREEDOM = 1359  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH



$F = +1.2048420E+01$   
 $R = +1.4854207E-01$   
 $I = +3.4710834E+00$   
 $N = 536$   
 $Y = (( +8.3294378E+01 ) + ( +1.4302153E-02 ) * X)$   
 SIGNIFICANCE OF F = SIGNIFICANT  
 SIGNIFICANCE OF R = SIGNIFICANT  
 SIGNIFICANCE OF I = SIGNIFICANT  
 DEGREES OF FREEDOM = 534  
 STORAGE CONDITIONS = AMB TEMP/AH  
 TEST CONDITIONS = AMB TEMP/AH



STAGE 1, WING 6, TP-H1043, HARDNESS SHORE A, INITIAL

Figure 22



TABLE 12

ANALYSIS OF COVARIANCE  
COMPARISON OF REGRESSION LINES

ANALYSIS OF COVARIANCE COMPARES REGRESSIONS WITH 2/6, TP-H1043, PARALLEL SH-4, LITL

SOURCE	D.F.	SUM XX	CORRECTED SUM SQUARES	SUM XY	SUM YY	REGRESSION COEFFICIENTS	D.F.	DEVIATIONS FROM REGRESSION SUM SQ.	MEAN SQ.
BETWEEN	1	0.21897E+07	9230.0	8561.7	1359	0.42425E-02	1	6521.3	4.7906
WITHIN	55	0.23249E+06	4754.4	5041.5	534	0.10302E-01	55	3013.0	5.6439
TOTAL	56								
POOLED	1895	0.25222E+07	14044	9642.5	1094	0.55609E-02	1894	9564.3	5.0498
DIFFERENCE BETWEEN SLOPES									
BETWEEN	1	0.21150E+06	-9712.0	108.00					
TOTAL	1896	0.27359E+07	9532.4	9751.5	1095				
DIFFERENTIAL ADJUSTED MEANS									
BETWEEN	1								
TOTAL	1896								

STEP 1:

COMPARISON BETWEEN RESIDUAL VARIANCES: F = 1.16 (DF = 524, 1359)

IF HOMOGENEITY OF RESIDUAL VARIANCES IS TRUE CONTINUE WITH STEP 2.

STEP 2:

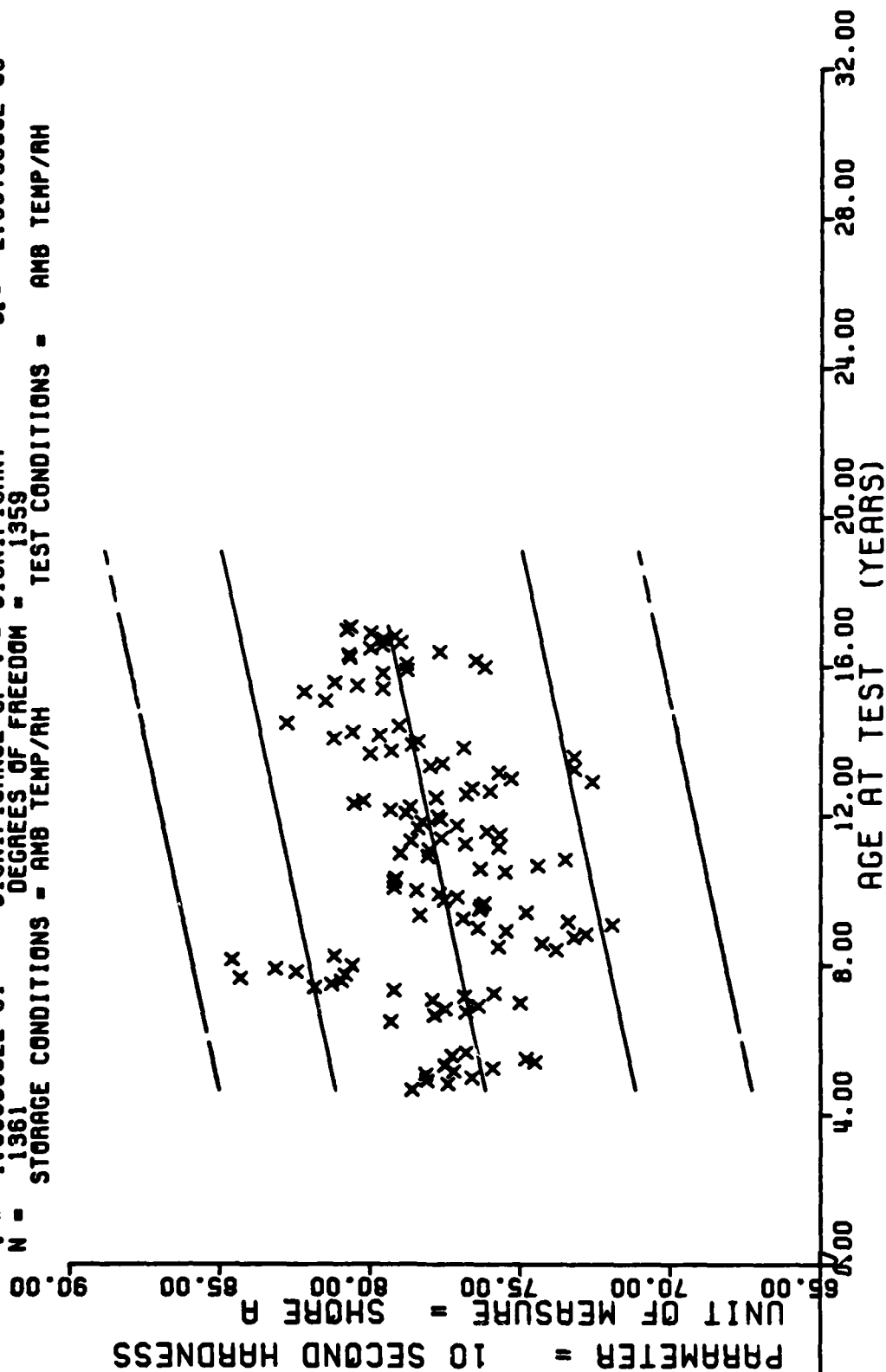
COMPARISON OF REGRESSION TREND LINE SLOPES: F = 5.10 (DF = 1, 1895)

COMPARISON OF REGRESSION TREND LINE DEVIATIONS: F = 30.50 (DF = 1, 1894)

STATISTICALLY DIFF.

STATISTICALLY DIFF.  
STATISTICALLY DIFF.

$Y = ((+7.4920583E+01) + (+2.1984429E-02) * X)$   
 $F = +1.2096331E+02$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma = +3.0856057E+00$   
 $R = +2.8589157E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_e = +1.9988875E-03$   
 $t = +1.0998332E+01$  SIGNIFICANCE OF t = SIGNIFICANT  $S_t = +2.9579063E+00$   
 $N = 1361$  DEGREES OF FREEDOM = 1359  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH



STAGE 1. WING 2. TP-M1043. HARDNESS SHORE A, 10-SECOND

Figure 23

PARAMETER = 10 SECOND HARDNESS

UNIT OF MEASURE = SHORE A

AGE AT TEST (YEARS)

STORAGE CONDITIONS = AMB TEMP/AM

TEST CONDITIONS = AMB TEMP/AM

STAGE 1. WING 6, TP-H1043, HARDNESS SHORE A, 10-SECOND

Figure 24

TABLE 13

ANALYSIS OF COVARIANCE  
COMPARISON OF REGRESSION LINES[illegible]

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP	AGE (MOS)	NR SAMP
56	5	92	2	120	20	148	15	185	25
58	5	93	4	121	40	149	9	186	20
59	10	94	7	123	20	150	5	187	10
60	5	95	4	124	15	151	5	190	10
61	15	96	3	126	6	152	5	191	10
62	10	98	1	127	3	153	5	192	10
63	10	99	1	128	12	155	5	193	50
64	10	101	10	130	6	156	10	194	10
65	10	102	11	131	11	158	30	195	35
66	10	103	15	132	19	159	5	196	25
67	11	105	10	133	25	160	5	197	10
68	5	106	5	134	14	161	15	198	10
78	2	107	15	135	17	162	5	199	5
80	4	108	20	136	11	164	5	200	15
81	1	109	15	137	21	165	10	201	5
82	2	110	15	138	3	166	25	202	15
83	1	111	10	139	9	167	10	203	10
84	1	112	20	140	15	168	15	204	20
85	5	113	15	141	9	169	5	205	15
86	4	114	20	142	18	170	10		
87	3	115	10	143	12	171	15		
88	1	116	20	144	8	173	10		
89	4	117	15	145	16	174	5		
90	4	118	40	146	18	181	10		
91	2	119	10	147	3	184	5		

STAGE 1. WING 2. TP-H1043, HARDNESS SHURE C. INITIAL

This sample size summary is applicable to figures 25 and 27

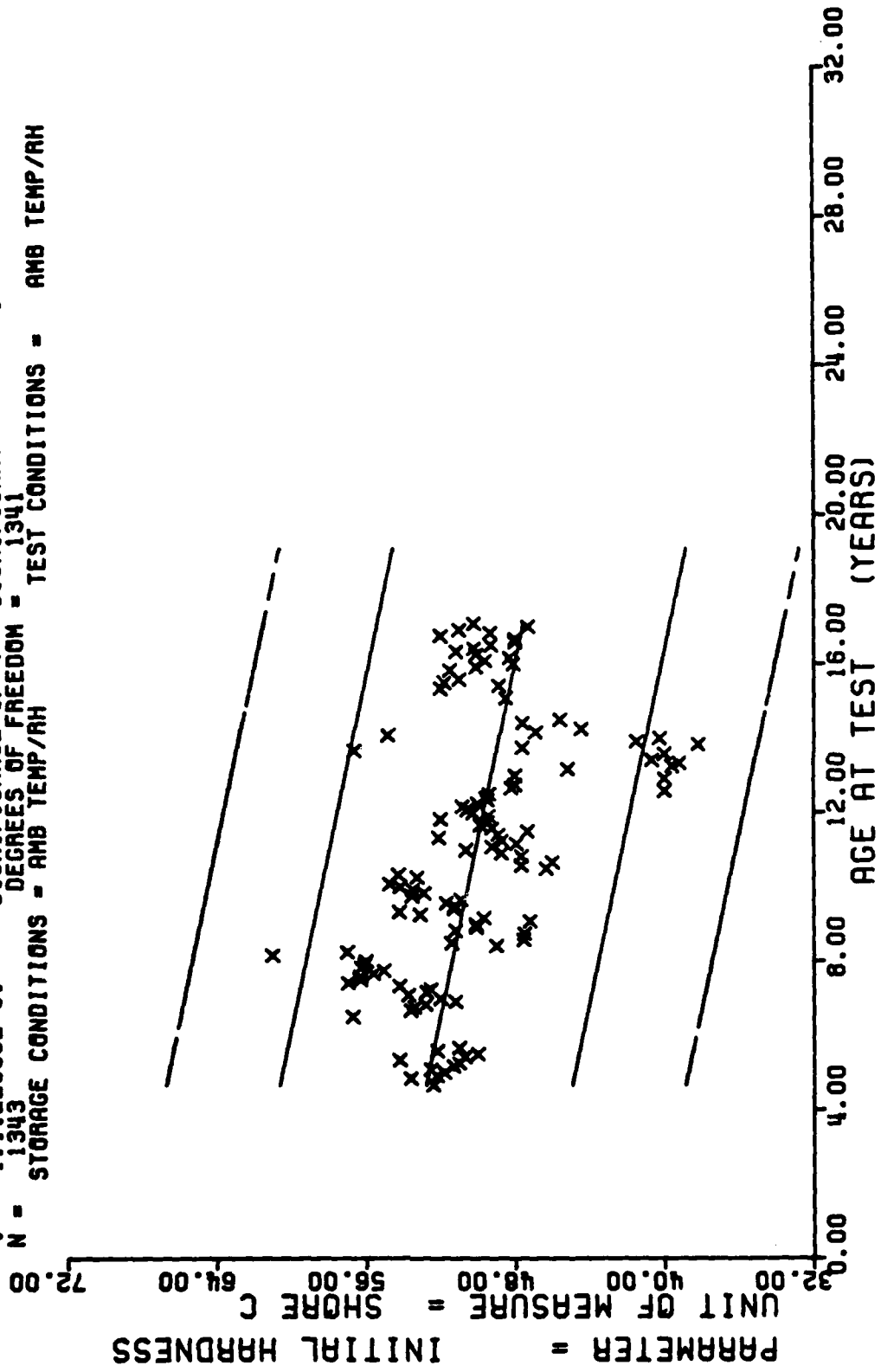
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

AGE (MDS)	NR SAMP	AGE (MDS)	NR SAMP
82	12	146	35
83	15	147	40
84	24		
85	27		
86	15		
87	18		
88	21		
89	21		
90	9		
91	9		
107	5		
108	10		
109	20		
110	25		
111	15		
112	15		
113	20		
114	15		
115	20		
140	5		
141	15		
142	15		
143	30		
144	50		
145	25		

STAGE 1. WING C. TP-H:043. HARDNESS SHORE C. INITIAL

This sample size summary is applicable to figures 26 and 28

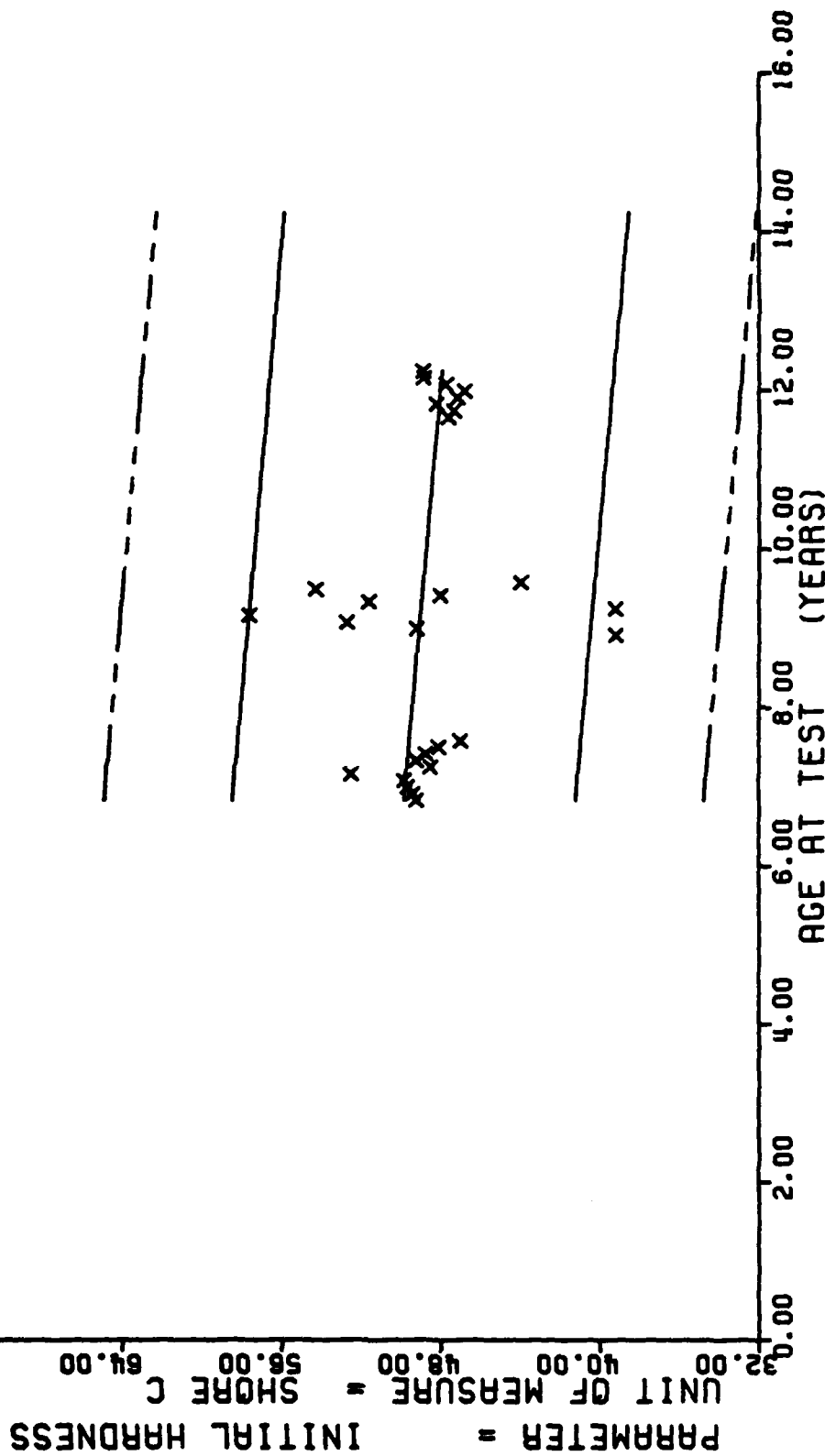
$Y = ((+5.4760067E+01) + (-3.5216968E-02) * X)$   
 $F = +1.2371085E+02$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma^2 = +4.8468345E+00$   
 $R = -2.9062164E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_e = +3.1662708E-03$   
 $t = +1.1122538E+01$  SIGNIFICANCE OF t = SIGNIFICANT  $S_e = +4.6393645E+00$   
 $N = 1343$  DEGREES OF FREEDOM = 1341  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH



STAGE 1. WING 2. TP-H1043. HARDNESS SHORE C. INITIAL

Figure 25

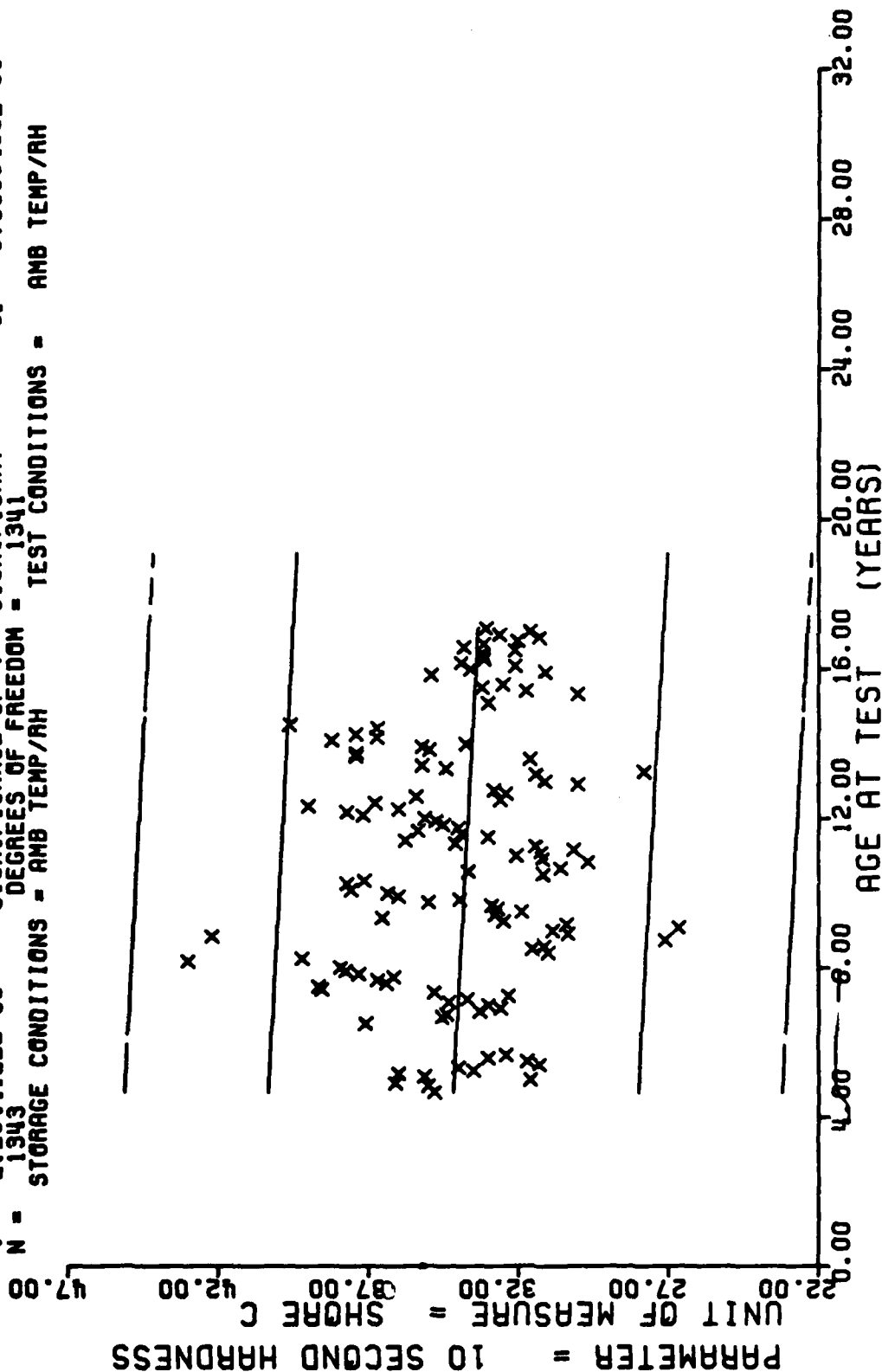
$Y = ((+5.2305275E+01) + (-2.9923191E-02) \times X)$   
 $F = +1.1732180E+01$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = -1.4729848E-01$  SIGNIFICANCE OF R = SIGNIFICANT  
 $t = +3.4252271E+00$  SIGNIFICANCE OF t = SIGNIFICANT  
 $N = 531$  DEGREES OF FREEDOM = 529  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH







$Y = ((+3.4481323E+01) + (-5.6467131E-03) \times X)$   
 $F = +5.1275630E+00$  SIGNIFICANCE OF F = SIGNIFICANT  
 $R = -6.1718078E-02$  SIGNIFICANCE OF R = SIGNIFICANT  
 $t = +2.2644122E+00$  SIGNIFICANCE OF t = SIGNIFICANT  
 $N = 1343$  DEGREES OF FREEDOM = 1341  
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH



STAGE 1, WING 2 TP-H1043, HARDNESS SHORE C, 10-SECOND

Figure 27

$Y = (( +4.1881612E+01 ) + ( -6.9309273E-02 ) \times X)$   
 $F = +1.5889372E+02$  SIGNIFICANCE OF F = SIGNIFICANT  $\sigma^2 = +3.3088612E+00$   
 $R = -4.7803612E-01$  SIGNIFICANCE OF R = SIGNIFICANT  $S_e = +5.0575856E-03$   
 $t = +1.2517736E+01$  SIGNIFICANCE OF t = SIGNIFICANT  $S_e = +2.9090514E+00$   
 $N = 531$  DEGREES OF FREEDOM = 529  
 STORAGE CONDITIONS = AMB TEMP/AMB TEST CONDITIONS = AMB TEMP/AMB

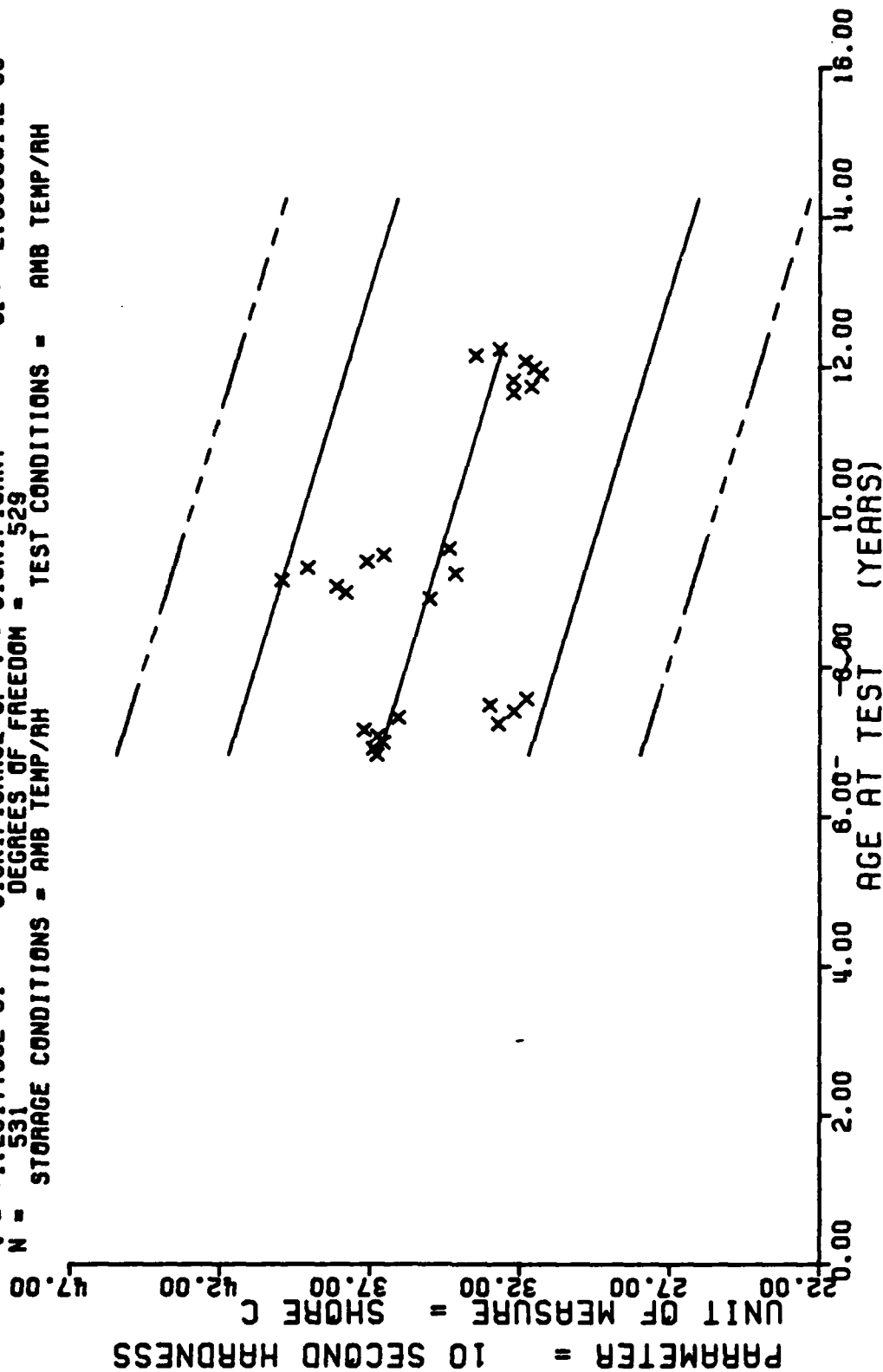


TABLE 15

1. The first of these is the fact that the Commission has not yet received any information from the Government of the United Kingdom regarding the progress of its investigation into the alleged activities of the British Intelligence Service in the United States.

ANALYSIS OF COMPARATIVE COUNTRIES REPORTS FOR 1962/63, 1963/64, 1964/65, 1965/66, 1966/67, 1967/68, 1968/69, 1969/70, 1970/71, 1971/72, 1972/73, 1973/74, 1974/75, 1975/76, 1976/77, 1977/78, 1978/79, 1979/80, 1980/81, 1981/82, 1982/83, 1983/84, 1984/85, 1985/86, 1986/87, 1987/88, 1988/89, 1989/90, 1990/91, 1991/92, 1992/93, 1993/94, 1994/95, 1995/96, 1996/97, 1997/98, 1998/99, 1999/00, 2000/01, 2001/02, 2002/03, 2003/04, 2004/05, 2005/06, 2006/07, 2007/08, 2008/09, 2009/10, 2010/11, 2011/12, 2012/13, 2013/14, 2014/15, 2015/16, 2016/17, 2017/18, 2018/19, 2019/20, 2020/21, 2021/22, 2022/23, 2023/24, 2024/25, 2025/26, 2026/27, 2027/28, 2028/29, 2029/30, 2030/31, 2031/32, 2032/33, 2033/34, 2034/35, 2035/36, 2036/37, 2037/38, 2038/39, 2039/40, 2040/41, 2041/42, 2042/43, 2043/44, 2044/45, 2045/46, 2046/47, 2047/48, 2048/49, 2049/50, 2050/51, 2051/52, 2052/53, 2053/54, 2054/55, 2055/56, 2056/57, 2057/58, 2058/59, 2059/60, 2060/61, 2061/62, 2062/63, 2063/64, 2064/65, 2065/66, 2066/67, 2067/68, 2068/69, 2069/70, 2070/71, 2071/72, 2072/73, 2073/74, 2074/75, 2075/76, 2076/77, 2077/78, 2078/79, 2079/80, 2080/81, 2081/82, 2082/83, 2083/84, 2084/85, 2085/86, 2086/87, 2087/88, 2088/89, 2089/90, 2090/91, 2091/92, 2092/93, 2093/94, 2094/95, 2095/96, 2096/97, 2097/98, 2098/99, 2099/00, 2100/01, 2101/02, 2102/03, 2103/04, 2104/05, 2105/06, 2106/07, 2107/08, 2108/09, 2109/10, 2110/11, 2111/12, 2112/13, 2113/14, 2114/15, 2115/16, 2116/17, 2117/18, 2118/19, 2119/20, 2120/21, 2121/22, 2122/23, 2123/24, 2124/25, 2125/26, 2126/27, 2127/28, 2128/29, 2129/30, 2130/31, 2131/32, 2132/33, 2133/34, 2134/35, 2135/36, 2136/37, 2137/38, 2138/39, 2139/40, 2140/41, 2141/42, 2142/43, 2143/44, 2144/45, 2145/46, 2146/47, 2147/48, 2148/49, 2149/50, 2150/51, 2151/52, 2152/53, 2153/54, 2154/55, 2155/56, 2156/57, 2157/58, 2158/59, 2159/60, 2160/61, 2161/62, 2162/63, 2163/64, 2164/65, 2165/66, 2166/67, 2167/68, 2168/69, 2169/70, 2170/71, 2171/72, 2172/73, 2173/74, 2174/75, 2175/76, 2176/77, 2177/78, 2178/79, 2179/80, 2180/81, 2181/82, 2182/83, 2183/84, 2184/85, 2185/86, 2186/87, 2187/88, 2188/89, 2189/90, 2190/91, 2191/92, 2192/93, 2193/94, 2194/95, 2195/96, 2196/97, 2197/98, 2198/99, 2199/00, 2200/01, 2201/02, 2202/03, 2203/04, 2204/05, 2205/06, 2206/07, 2207/08, 2208/09, 2209/10, 2210/11, 2211/12, 2212/13, 2213/14, 2214/15, 2215/16, 2216/17, 2217/18, 2218/19, 2219/20, 2220/21, 2221/22, 2222/23, 2223/24, 2224/25, 2225/26, 2226/27, 2227/28, 2228/29, 2229/30, 2230/31, 2231/32, 2232/33, 2233/34, 2234/35, 2235/36, 2236/37, 2237/38, 2238/39, 2239/40, 2240/41, 2241/42, 2242/43, 2243/44, 2244/45, 2245/46, 2246/47, 2247/48, 2248/49, 2249/50, 2250/51, 2251/52, 2252/53, 2253/54, 2254/55, 2255/56, 2256/57, 2257/58, 2258/59, 2259/60, 2260/61, 2261/62, 2262/63, 2263/64, 2264/65, 2265/66, 2266/67, 2267/68, 2268/69, 2269/70, 2270/71, 2271/72, 2272/73, 2273/74, 2274/75, 2275/76, 2276/77, 2277/78, 2278/79, 2279/80, 2280/81, 2281/82, 2282/83, 2283/84, 2284/85, 2285/86, 2286/87, 2287/88, 2288/89, 2289/90, 2290/91, 2291/92, 2292/93, 2293/94, 2294/95, 2295/96, 2296/97, 2297/98, 2298/99, 2299/00, 2300/01, 2301/02, 2302/03, 2303/04, 2304/05, 2305/06, 2306/07, 2307/08, 2308/09, 2309/10, 2310/11, 2311/12, 2312/13, 2313/14, 2314/15, 2315/16, 2316/17, 2317/18, 2318/19, 2319/20, 2320/21, 2321/22, 2322/23, 2323/24, 2324/25, 2325/26, 2326/27, 2327/28, 2328/29, 2329/30, 2330/31, 2331/32, 2332/33, 2333/34, 2334/35, 2335/36, 2336/37, 2337/38, 2338/39, 2339/40, 2340/41, 2341/42, 2342/43, 2343/44, 2344/45, 2345/46, 2346/47, 2347/48, 2348/49, 2349/50, 2350/51, 2351/52, 2352/53, 2353/54, 2354/55, 2355/56, 2356/57, 2357/58, 2358/59, 2359/60, 2360/61, 2361/62, 2362/63, 2363/64, 2364/65, 2365/66, 2366/67, 2367/68, 2368/69, 2369/70, 2370/71, 2371/72, 2372/73, 2373/74, 2374/75, 2375/76, 2376/77, 2377/78, 2378/79, 2379/80, 2380/81, 2381/82, 2382/83, 2383/84, 2384/85, 2385/86, 2386/87, 2387/88, 2388/89, 2389/90, 2390/91, 2391/92, 2392/93, 2393/94, 2394/95, 2395/96, 2396/97, 2397/98, 2398/99, 2399/00, 2400/01, 2401/02, 2402/03, 2403/04, 2404/05, 2405/06, 2406/07, 2407/08, 2408/09, 2409/10, 2410/11, 2411/12, 2412/13, 2413/									
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iii

COMPARISON BETWEEN RESIDUAL VARIANCES:  $F = 1.56 (F_{1, 529})$

4. COEFFICIENT OF RESIDUAL VARIATION IS FOUR CONTINUED WITH STEP 2.

2712

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This report contains propellant test results from cartons of TP-H1043 propellant representing selected batches used in the aft closure of First Stage Minuteman Motors. Data from TP-H1043 propellant obtained from the aft closures of the LGM-30 A, B, F, and G Motors are reported in regression analyses for the sixth time and the fifth time using the G085 Computer System. In previous reports, A, B, F, and G data were combined for statistical analysis. For this report, a separate statistical analysis was made for A and B (wing 2) and		

F and G (wing 6). The statistical analysis includes regressions and covariance analysis. Testing was accomplished in accordance with MMWRBA Project M8293C.

An analysis of all parameters indicate that no significant degradation is anticipated for at least two years past the oldest data point.

Each point on the regression plot represents all samples at that particular age. The number of samples at each point is indicated on the sample size summary sheet on the page accompanying each regression or group of regression plots. The data range at any age can be found by suitable inquiry of the G085 system.

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